

EMERGENCY PROCEDURE TABS

ENGINE FIRE

1

NAVAIR 01-90KDB-1B
**NATOPS PILOT'S
 POCKET CHECKLIST**

**T-34B
 AIRCRAFT**

THIS PUBLICATION SUPERSEDES
 NAVAIR 01-90KDB-1B DATED
 1 DECEMBER 1979

ISSUED BY AUTHORITY OF
 THE CHIEF OF NAVAL OPERATIONS
 AND UNDER THE DIRECTION OF
 THE COMMANDER, NAVAL AIR
 SYSTEMS COMMAND

PERFORMANCE DATA INDEX

LIFT TIME

- 2 Emer Landing Pattern
- 3 T/O-Ground Run
- 4 T/O-50' Obstacle
- 5 Landing Distance
- 6 Fouled Deck Range
- 7 Fouled Deck Endurance

ENGINE FAILURE

2

FIRE IN FLIGHT
 SMOKE/FUMES
 PROPELLER FAIL

3

ELECTRICAL FAIL
 LOST PLANE

4

DAMAGED A/C
 BAILOUT

5

DITCHING

6

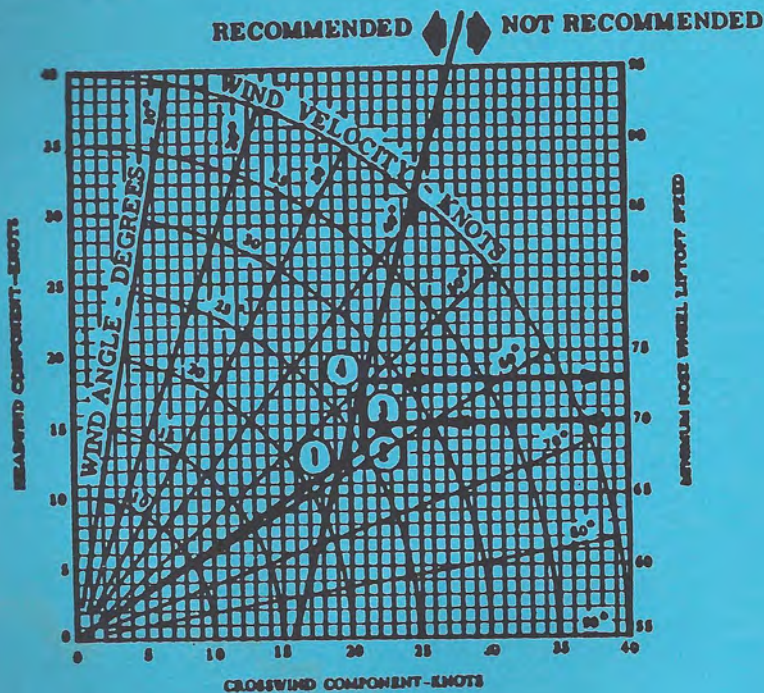
T.O./LDG EMER

7

1 MARCH 1981

TAKEOFF CROSSWIND CHART

MODEL T-34B

ENGINE (1)
NO.0-470-4

SAMPLE PROBLEM:

KNOWN:
TAKEOFF RUNWAY - 3
WIND - 67^{kt}/31 KNOTS

DETERMINE:
IF TAKEOFF IS RECOMMENDED AT
TAKEOFF SPEED OF 70 KNOTS IAS.

NOTE

- MAXIMUM NOSEWHEEL LIFT-OFF SPEED IS 90 KNOTS IAS.
- ENTER CHART AT MAXIMUM GUST VELOCITY.

DATA BASED ON: FLIGHT TEST

FUEL GRADE: 80/87
FUEL DENSITY: 6 LB/GAL

SOLUTION:

- ① WIND ANGLE IS 76°. 26° - 90° WIND ANGLE TO RUNWAY HEADING.
- ② AT WIND VELOCITY OF 35 KNOTS AND 60° WIND ANGLE TO RUNWAY HEADING, FIND CROSSWIND COMPONENT OF 21 KNOTS AND HEADWIND COMPONENT OF 13 KNOTS.
- ③ PROCEED VERTICALLY TO PREDICTED TAKEOFF SPEED OF 70 KNOTS IAS AND DETERMINE TAKEOFF AS NOT RECOMMENDED.
- ④ CONTINUE VERTICALLY AND DETERMINE THAT TAKEOFF SPEED MUST BE 73 KNOTS IAS BEFORE TAKEOFF IS RECOMMENDED.

NAVAIR 01-90KDB-1B

**NATOPS PILOT'S
POCKET CHECKLIST
T-34B**

AIRCRAFT MAR 30 1982

AIMD JUSTIFICATION

COPY NO. 1

LOCATION Flying Club

THIS PUBLICATION SUPERSEDES
NAVAIR 01-90KDB-1B DATED DATE
1 DECEMBER 1978

1 Mar 81

CHANGE DATE

ISSUED BY AUTHORITY OF THE CHIEF OF NAVAL OPERATIONS
AND UNDER THE DIRECTION OF THE COMMANDER,
NAVAL AIR SYSTEMS COMMAND

1 MARCH 1981

Reproduction for nonmilitary use of the information or illustrations contained in this publication is not permitted without specific approval of the Commander, Naval Air Systems Command.

LIST OF EFFECTIVE PAGES

Note: Text affected by current change indicated by vertical line in outer margin.

TOTAL NUMBER OF PAGES IN THIS PUBLICATION
IS 63 CONSISTING OF THE FOLLOWING:

<i>Page No.</i>	<i>Issue</i>
Cover	Original
Inside Front Cover	Original
Title	Original
A	Original
B/(C blank)	Original
1-2	Original
3/(4 blank)	Original
5-6	Original
7/(8 blank)	Original
9/(10 blank)	Original
11-18	Original
19/(20 blank)	Original
21-22	Original
23/(24 blank)	Original
25/(26 blank)	Original
27/(28 blank)	Original
29-44	Original
45/(46 blank)	Original
47-48	Original
49/(50 blank)	Original
51-56	Original
Inside Back Cover	Original

ADDITIONAL COPIES:

Additional copies of this manual and changes thereto may be procured by submitting Form DD 1348 to NPFC Philadelphia in accordance with Introduction to Navy Stocklist of Publications and Forms NAVSUP Publication 2002 (S/N 0535-LP-004-0001).

INTERIM CHANGE SUMMARY

The following Interim Changes have been canceled or previously incorporated in this manual:

INTERIM CHANGE NUMBER(S)	REMARKS/PURPOSE
1 thru 8	Previously incorporated

The following Interim Changes have been incorporated in this Change /Revision

INTERIM CHANGE NUMBER	REMARKS/PURPOSE
9	Low altitude engine failure

Interim Changes Outstanding—To be maintained by the custodian of this manual

INTERIM CHANGE NUMBER	ORIGINATOR/DATE (or DATE/TIME GROUP)	PAGES AFFECTED	REMARKS/PURPOSE

DURING START**ENGINE FIRE**

- *1. Mixture..... IDLE CUTOFF
- *2. Fuel shutoff valve handleOFF
- *3. Throttle..... FULL FORWARD
- *4. Continue cranking to clear engine,
attempting a start.

NEW

If no start:

- *5. Ignition.....OFF
- *6. BatteryOFF
- *7. ABANDON AIRCRAFT.

AFTER START ON GROUND

- *1. Mixture..... IDLE CUTOFF
- *2. Fuel shutoff valve handleOFF
- *3. Throttle..... FULL FORWARD
- *4. Ignition.....OFF
- *5. BatteryOFF
- *6. ABANDON AIRCRAFT.

NEW

CONTINUED***Memory items**

EMERGENCY LANDING PATTERN

TO BE USED FOR:

- ENGINE FAILURE OR MALFUNCTION
- PRECAUTIONARY EMERGENCY LANDING
- SIMULATED ENGINE FAILURE

① DESCENT

- A/S—90 KIAS
- GEAR—UP
- Flaps—UP
- Canopy—CLOSED
(open prior to high key)
- Prop—HIGH PITCH*

② HIGH KEY (1,500 FEET AGL)

- Over intended point of landing, 90 KIAS
- Begin turn to Low Key, Gear—DOWN for prepared surfaces. Transition to 85 KIAS. Gear—UP for unprepared surfaces or water. Maintain 90 KIAS.

③ LOW KEY (1,000 FEET AGL)

- 85 KIAS (if gear down), 90 KIAS (if gear up), wingtip distance abeam intended point of landing.

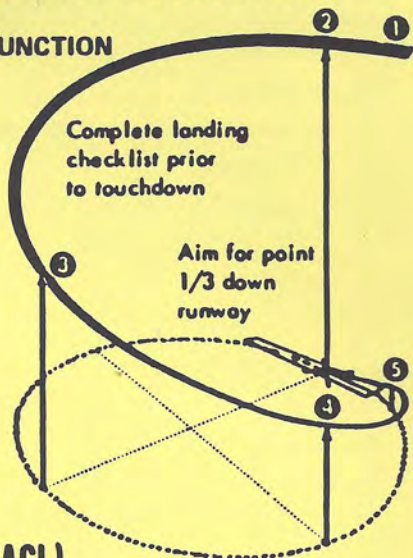
④ 90-DEGREE (500-600 FEET AGL)

- Complete Landing Checklist
- Flaps—AS DESIRED. Adjust to 75 KIAS with flaps down.

⑤ FINAL

- 800 feet straightaway
- 200 feet AGL
- Canopy—BLOW OPEN*

*Only for actual engine failure



HIGH ALTITUDE / PARTIAL

ENGINE FAILURE

- *1. Assume a safe flight attitude.
- *2. Select the best available landing area and turn to intercept the emergency landing pattern at the maximum altitude practicable. If power is available, climb to an altitude from which the aircraft can glide to a high key position.
- *3. Gear and flaps AS DESIRED
(aircraft clean will extend a glide).
- *4. Fuel boost pump ON
- *5. Fuel shutoff valve handle ON
- *6. Mixture RICH
- *7. Propeller FULL INCREASE
- *8. Throttle FULL FORWARD
- *9. Ignition ON, BOTH

If engine still not running or altitude cannot be maintained, activate emergency fuel system as follows:

- *10. Emergency fuel switch ON
- *11. Mixture IDLE CUTOFF

If engine does not start:

- *12. Fuel shutoff valve handle OFF
- *13. Propeller AS REQUIRED
- *14. Ignition OFF
- *15. Emergency fuel switch OFF
- *16. Gear AS REQUIRED
- *17. Flaps AS DESIRED
- 18. Transmit appropriate radio call
- *19. Battery OFF
- *20. Generator OFF
- *21. Canopy BLOWN
- *22. Harness LOCKED

*Memory items

NEW NEW NEW NEW NEW NEW NEW NEW NEW NEW NEW NEW NEW NEW NEW

LOW ALTITUDE ENGINE FAILURE

N
E
W

If engine fails at or below 1000 feet AGL:

- *1. Assume a safe gliding attitude.
- *2. Select the best available landing area and turn to intercept the emergency landing pattern at the maximum altitude practicable.
- *3. Emergency fuel switchON

If power is not regained, execute the following prior to landing:

- *4. GearAS DESIRED
- *5. FlapsAS DESIRED
- *6. Fuel shutoff valve handleOFF
- *7. BatteryOFF
- *8. CanopyBLOWN
- *9. HarnessLOCKED

N
E
W
N
E
W
N
E
W
N
E
W
N
E
W

*Memory items

GROUND RUN TAKEOFF

ZERO FLAPS — HARD SURFACE RUNWAY
GROUND ROLL DISTANCES

		PRESSURE ALTITUDE — FEET								
	RUNWAY TEMP. °C	SEA LEVEL	1000	2000	3000	4000	6000	WIND VELOCITY		
GROSS WEIGHT 2475 POUNDS	0	615	675	745	815	910	1170	} 0 KNOTS		
	15	720	785	845	970	1085	1380			
	30	810	910	1015	1140	1270	1600			
	GROSS WEIGHT 2475 POUNDS	0	360	400	450	500	570	740	} 20 KNOTS	
		15	425	490	550	610	695	900		
		30	495	570	655	725	820	1040		
		GROSS WEIGHT 2475 POUNDS	0	280	315	360	390	460	600	} 30 KNOTS
			15	345	390	425	500	565	725	
			30	395	460	530	590	650	850	
GROSS WEIGHT 2775 POUNDS			0	890	950	1025	1115	1230	1550	} 0 KNOTS
			15	980	1070	1170	1305	1445	1850	
			30	1105	1230	1355	1520	1695	2170	
	GROSS WEIGHT 2775 POUNDS		0	550	600	655	710	795	1010	} 20 KNOTS
			15	596	680	715	840	950	1210	
			30	705	800	870	990	1110	1400	
		GROSS WEIGHT 2775 POUNDS	0	440	490	540	580	630	830	} 30 KNOTS
			15	480	560	600	680	775	990	
			30	580	640	705	810	910	1160	
GROSS WEIGHT 2975 POUNDS			0	1060	1140	1230	1350	1490	1880	} 0 KNOTS
			15	1185	1290	1420	1575	1750	2240	
			30	1315	1500	1630	1840	2050	2620	
	GROSS WEIGHT 2975 POUNDS		0	655	710	780	860	970	1210	} 20 KNOTS
			15	730	830	935	1020	1145	1450	
			30	850	970	1070	1195	1340	1640	
		GROSS WEIGHT 2975 POUNDS	0	550	590	635	700	785	1005	} 30 KNOTS
			15	595	660	750	830	940	1200	
			30	690	780	875	985	1100	1370	

Ground Run Takeoff

ENGINE FIRE

FIRE IN FLIGHT

- *1. Mixture IDLE CUTOFF
- *2. Fuel shutoff valve handle OFF
- *3. Throttle CLOSED
- *4. Ignition OFF
- *5. Battery OFF
- *6. Generator OFF
- 7. Do not attempt restart.
- 8. Execute emergency landing or bail out if fire persists.

ELECTRICAL FIRE

- *1. Battery OFF
- *2. Generator OFF
- 3. All circuit breakers PULLED
- 4. All radio/electrical equipment OFF

If fire persists:

- 5. Make emergency landing or bail out.

To isolate faulty circuit:

- 6. Generator circuit breaker IN
- 7. Generator ON
(if faulty OFF)
- 8. Battery ON
- 9. Check each necessary circuit one at a time by pushing IN circuit breaker and turning ON radio/electrical equipment it services.
- 10. Secure unnecessary radio/electrical equipment to conserve battery if generator is secured.

CONTINUED

*Memory items

WING FIRE

A fire in the wing could be caused by fuel leakage and/or defective electrical wiring. Perform the following procedure:

- *1. Battery and generator switchesOFF
- *2. Attempt to extinguish the fire by slipping aircraft away from the fire.
- *3. If fire does not extinguish or is obviously fed by aircraft fuelBAIL OUT

NEW NEW NEW

FUSELAGE FIRE

- *1. Reduce airspeed
- *2. Canopy CLOSED
- *3. Cockpit air handles FULL OUT
(air shut off)
- *4. Battery and generator switchesOFF
- *5. If fire persistsBAIL OUT OR LAND

SMOKE/
FUMES

ELIMINATION OF SMOKE

SMOKE/FUMES

- *1. Airspeed REDUCE
(to minimize spreading of possible fire)
- *2. CanopyOPEN
- *3. Cockpit air handles FULL OUT
- 4. Determine source of smoke and execute appropriate emergency procedures.

CARBON MONOXIDE

If carbon monoxide contamination is suspected:

- *1. CanopyOPEN
- *2. Cockpit air handles FULL OUT

*Memory items

FUEL LEAK/FUEL FUMES

NEW

Check fuel system for secondary indications and proceed as follows:

- *1. Maintain present airspeed.
- *2. Land as soon as practicable.
- *3. CanopyOPEN
- *4. Cockpit air handles FULL OUT
- *5. Battery and generator switchesOFF
- *6. Utilize landing gear emergency extension system.
- *7. Accomplish landing, clear runway, secure engine and ABANDON AIRCRAFT.

NEW

PROP
FAIL

PROPELLER FAILURE

- *1. Adjust throttle to maintain safe flight while minimizing overspeed.
- *2. Climb to put load on propeller.
- *3. Manipulate propeller control in an attempt to restore governing.
- *4. Land as soon as possible.

NEW
NEW
NEW

*Memory items

50-FOOT OBSTACLE TAKEOFF

ZERO FLAPS — HARD SURFACE RUNWAY
DISTANCE TO CLEAR 50-FOOT OBSTACLE

		PRESSURE ALTITUDE — FEET						WIND VELOCITY	
	RUNWAY TEMP. °C	SEA LEVEL	1000	2000	3000	4000	6000		
GROSS WEIGHT 2475 POUNDS	0	780	870	920	1180	1400	2050	} 0 KNOTS	
	15	950	1090	1280	1520	1820	2690		
	30	1145	1375	1640	1970	2370	3480		
	}	0	450	550	660	740	890	1340	} 20 KNOTS
		15	580	680	810	970	1170	1800	
		30	725	880	1040	1300	1570	2360	
	}	0	330	420	520	590	720	900	} 30 KNOTS
		15	460	550	650	800	960	1320	
		30	570	720	850	1070	1280	1960	
GROSS WEIGHT 2775 POUNDS	0	1000	1140	1300	1540	1800	2640	} 0 KNOTS	
	15	1220	1410	1670	1970	2350	3460		
	30	1490	1780	2110	2540	3060	4540		
	}	0	645	720	840	980	1170	1750	} 20 KNOTS
		15	770	990	1060	1310	1570	2350	
		30	970	1160	1370	1680	2030	3220	
	}	0	510	580	670	800	960	1440	} 30 KNOTS
		15	610	800	870	1080	1290	1950	
		30	785	960	1120	1380	1660	2720	
GROSS WEIGHT 2975 POUNDS	0	1180	1340	1540	1790	2080	3040	} 0 KNOTS	
	15	1425	1640	1930	2280	2710	4030		
	30	1740	2060	2430	2920	3530	5320		
	}	0	740	860	990	1160	1360	2040	} 20 KNOTS
		15	900	1080	1250	1490	1800	2800	
		30	1125	1350	1620	1995	2370	3800	
	}	0	600	700	810	910	1120	1680	} 30 KNOTS
		15	730	880	1020	1220	1480	2320	
		30	925	1100	1320	1590	1970	3200	

50-Foot Obstacle Takeoff

DC POWER (GENERATOR)

ELECTRICAL FAILURE

1. GeneratorOFF
(if warning light is on or voltage exceeds 30 volts)
2. Nonessential electric equipmentOFF
(to conserve battery)

If complete electrical system failure:

3. GeneratorOFF
4. BatteryOFF

AC POWER (INVERTER)

1. Inverter switchSTANDBY
(if inverter out light is on)

LOST PLANE

LOST PLANE

1. Confess.
2. Communicate.
3. Climb.
4. Conserve.
5. Comply with enroute procedures.
6. Know any peculiar local area procedures.

*Memory items

LANDING DISTANCE

FULL FLAPS — HARD SURFACE RUNWAY
GROUND ROLL DISTANCE — FEET

		HEAD-WIND VELOCITY					PRESSURE ALTITUDE		
		RUNWAY TEMP. °C	0 KN.	10 KN.	20 KN.	30 KN.		40 KN.	
GROSS WEIGHT 2475 POUNDS	0	345	240	175	130	105	} SEA LEVEL		
	15	360	245	180	135	110			
	30	375	260	190	145	115			
	GROSS WEIGHT 2775 POUNDS	0	375	260	190	145	115	} 3000 FEET	
		15	390	270	200	155	125		
		30	400	280	210	160	130		
		GROSS WEIGHT 2975 POUNDS	0	405	290	215	165	130	} 4000 FEET
			15	420	295	220	170	135	
			30	435	305	230	180	145	
GROSS WEIGHT 2475 POUNDS			0	390	270	200	155	125	} SEA LEVEL
			15	400	280	210	160	130	
			30	420	295	220	170	135	
	GROSS WEIGHT 2775 POUNDS		0	420	295	220	170	135	} 3000 FEET
			15	435	305	230	180	145	
			30	455	325	245	190	155	
		GROSS WEIGHT 2975 POUNDS	0	460	330	250	200	160	} 4000 FEET
			15	475	345	260	205	165	
			30	490	355	270	210	170	
GROSS WEIGHT 2475 POUNDS			0	410	295	220	170	140	} SEA LEVEL
			15	430	305	230	180	145	
			30	450	320	240	190	155	
	GROSS WEIGHT 2775 POUNDS		0	450	320	240	190	155	} 3000 FEET
			15	470	340	255	205	165	
			30	490	355	270	210	170	
		GROSS WEIGHT 2975 POUNDS	0	490	355	270	210	170	} 4000 FEET
			15	495	360	275	215	175	
			30	530	390	300	240	195	

Landing Distance

AIRBORNE

DAMAGED AIRCRAFT

1. If aircraft is controllable, climb to at least 5000 feet.
2. Communicate – state difficulty, request visual inspection.
3. Check flight characteristics:
 - a. Gear and flaps – 85 knots
 - b. Gear down, flaps up – 90 knots
4. Fly wide approach, maintaining 10 knots above minimums obtained during flight tests.

BAILOUT

BAILOUT

- *1. Make radio distress call, time permitting.
- *2. Warn other pilot to prepare to bail out and receive acknowledgment.
- *3. Reduce airspeed as much as practicable, with flaps extended, trim slightly nosedown, and head for uninhabited area.
- *4. Radio cords.....DISCONNECT
- *5. Emergency canopy open.....PULL
- *6. Seat.....FULL UP
- *7. Parachute straps.....TIGHT
- *8. Harness.....RELEASE
- *9. Assume crouch position on seat.
- *10. Dive for trailing edge of wing.
- *11. When clear of aircraft.....PULL D-RING

N-1

FOULED DECK RANGE

CLEAN AIRPLANE

RANGE CLIMB SCHEDULE		
ALTITUDE (FEET)	CLIMB IAS	CRUISE IAS
SEA LEVEL	91	105
5000	85	105
10,000	79	105

DESCENT AIRSPEED 115 KIAS

FLIGHT ALTITUDE (FEET)	FUEL ON BOARD (POUNDS)	DESCEND OR CLIMB TO OPTIMUM ALTITUDE (FEET)	MAXIMUM RANGE (MILES)	START DESCENT MILES FROM DESTINATION	TO ARRIVE AT SEA LEVEL WITH FUEL ON BOARD (POUNDS)
SEA LEVEL	250	5000	670	8	23
SEA LEVEL	200	5000	531	7	20
SEA LEVEL	150	5000	390	7	18
5000	250	5000	686	8	23
5000	200	5000	547	7	20
5000	150	5000	401	7	18
10,000	250	5000	682	8	23
10,000	200	5000	542	7	20
10,000	150	5000	396	7	18

Fouled Deck Range

DITCHING**DITCHING**

1. Plan to touch down before all fuel is exhausted, to have power for controlled approach.
2. Make radio distress call.
3. Radio cords.....DISCONNECT
4. Harness.....LOCK
5. Parachute straps.....UNBUCKLE
6. Emergency canopy open.....PULL
7. Landing gear.....UP
8. Flaps.....DOWN
9. Battery.....OFF
10. Make normal approach with power, if possible. Approach stall attitude at a speed under which full control of aircraft can be maintained. Plan landing direction as follows:

Calm sea – Into wind

Moderate swells – Parallel to swells

High swells (25 knots of wind or more) – Into wind, attempting to land on upwind side of swell.

11. Release safety belt **ONLY** after aircraft has come to full stop.
12. **ABANDON AIRCRAFT.**

***Memory items**

FOULED DECK ENDURANCE

CLEAN AIRPLANE

ENDURANCE CLIMB SCHEDULE	
ALTITUDE	IAS
SEA LEVEL	91
5000	85
10,000	79

LOITER AIRSPEED 70 KIAS

DESCENT 93 KIAS

FUEL ON BOARD (POUNDS)	FLIGHT ALT (FEET)	ENDURANCE (MINUTES)	DESCEND OR CLIMB TO (FEET)	DESCEND WHEN FUEL IS (POUNDS)	FUEL REMAINING AT SEA LEVEL (POUNDS)
250	SEA LEVEL	440	5000	27	23
250	5000	451	5000	27	23
250	10,000	448	5000	27	23
200	SEA LEVEL	348	5000	24	20
200	5000	360	5000	24	20
200	10,000	356	5000	24	20
150	SEA LEVEL	261	5000	22	18
150	5000	268	5000	22	18
150	10,000	265	5000	22	18

Fouled Deck Endurance

ABORTED TAKEOFF

T.O./LDG EMER.

- *1. Throttle..... CLOSED
- *2. Brakes.....APPLIED

If unable to stop on runway:

- *3. Canopy BLOW OPEN
- *4. Mixture..... IDLE CUTOFF
- *5. Fuel shutoff valve handleOFF
- *6. Ignition.....OFF
- *7. BatteryOFF
- *8. ABANDON AIRCRAFT after it stops.

EMERGENCY EXTENSION

- 1. Landing gear circuit breakerOUT
- 2. Landing gear handle DOWN
- 3. Clutch knobUNLOCK
- 4. Push clutch knob DOWN to engage crank.
- 5. Crank gear..... DOWN
(approximately 37 turns)
- 6. Check gear.....DOWN AND LOCKED

*Memory items

GENERAL

LANDING EMERGENCY

If the gear cannot be lowered successfully, proceed with the emergency procedures for the appropriate gear malfunction.

CAUTION

- If an unsafe gear indication existed and the gear have been successfully lowered, do not attempt to raise the gear.
- Raising the gear after a malfunction could cause further damage.

NEW
NEW

GEAR UP LANDING

1. Make normal approachFULL FLAPS
2. Emergency canopy open handlePULL
3. Harness LOCKED

NEW

After touchdown:

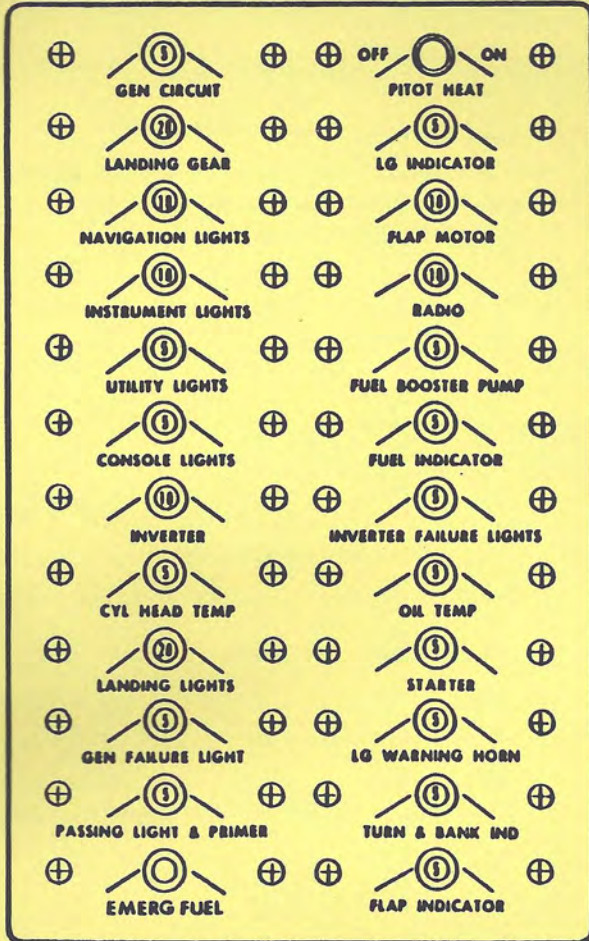
4. Mixture IDLE CUTOFF
5. Fuel shutoff valve handleOFF
6. BatteryOFF
7. ABANDON AIRCRAFT as soon as it stops.

NEW

**LDG
EMER**

CONTINUED

COCKPIT CIRCUIT BREAKERS FORWARD COCKPIT ONLY



Cockpit Circuit Breakers

ONE MAIN GEAR RETRACTED

1. Have gear position checked visually by another pilot or by the tower on a flyby, if possible.
2. If verified that one gear is not fully extended and an attempt to retract it is unsuccessful, execute normal approach with full flaps and power on to reduce landing speed, carrying the wing slightly lower on the down and locked side.
3. Emergency canopy open handle PULL
4. Touch down smoothly on the down and locked gear. Hold opposite wing up with aileron as long as possible after nosewheel touches down.
5. When wingtip strikes the ground, apply maximum opposite brake pressure.
6. As soon as aircraft stops:
 - Mixture IDLE CUTOFF
 - Fuel shutoff valve handle OFF
 - Battery OFF
7. ABANDON AIRCRAFT.

NOSE GEAR RETRACTED

1. Make a normal approach
2. Emergency canopy open handle PULL
3. After touching main wheel down, hold nose up as long as possible with full nose down elevator trim and full backstick.

Before nose settles onto ground:

4. Mixture IDLE CUTOFF
5. Fuel shutoff valve handle OFF
6. Battery OFF
7. ABANDON AIRCRAFT as soon as it stops.

CONTINUED ➤

NOSE GEAR MALFUNCTION

1. Reduce airspeed; lower gear and flaps.
2. Assume slow flight -- 70 knots.
3. Make gentle pitching oscillations (use centrifugal force to swing nose gear into down position).
4. When landing, lower nosewheel to runway gently.
5. Use forward stick to keep nosewheel firmly on runway.

FLAT TIRE

1. Touch down well over opposite side of runway to allow room for a swerve and hold directional control with opposite brake.
2. Avoid hard applications of brake.
3. After landing with a flat tire, perform the Secure Checklist when the aircraft comes to a complete stop and have the aircraft towed clear of the landing area.
4. Do not taxi in with a flat tire.

BRAKE FAILURE

If no brake pressure was evident during landing pattern brake check, land aircraft as short as possible using full flaps to shorten landing roll. After touchdown, secure the engine. When the aircraft comes to a complete stop, complete the remaining items on the Secure Checklist and have the aircraft towed clear of the landing area.

CONTINUED ➤

HARD LANDINGS

If on the runway:

- 1. Runway permitting, execute a full stop.**
- 2. Do not attempt to taxi the aircraft.**

If airborne:

- 3. Have landing gear checked visually by another pilot or by the tower on a flyby, if possible.**
- 4. If the check reveals no visible damage, execute a normal full flap landing and proceed as in steps 1 and 2.**
- 5. If visual damage is confirmed, execute appropriate emergency procedure.**

**NORMAL PROCEDURES
TABLE OF CONTENTS**

PREFLIGHT INSPECTION29
PRESTART37
STARING ENGINE38
PRETAKEOFF39
TAKEOFF43
LANDING44

PREFLIGHT INSPECTION

Items marked with an asterisk (*) are to be checked prior to the first flight of the day and may be omitted for subsequent flights that day.

FORWARD COCKPIT

- 1. Boost pumpOFF
- 2. Fuel shutoff valve handleOFF
- 3. Trim tabs0 DEGREES
- 4. MixtureIDLE CUTOFF
- 5. IgnitionOFF
- 6. Landing gear handleDOWN
- 7. Emergency landing gear retract switchOFF (WIRED)
- 8. Accelerometer limits+4.0 TO -2.0
- 9. Emergency fuel switchOFF
- 10. BatteryOFF
- 11. ControlsUNLOCKED
- *12. Lap belt and shoulder harness:
 - a. WebbingCHECK FOR CUTS
 - b. Release buckleOPERATION

NEW
NEW
NEW
NEW
NEW
NEW

- 13. Parachute/lanyardCONDITION/
CONNECTED.
- *14. Cockpit/equipmentCLEAN/
SECURED AND
STOWED.
- *15. Canopy and windshieldCHECK FOR
CRAZING.
- *16. Canopy sealSECURITY.
- *17. Canopy actuating mechanism
(handles, rollers, and tracks)CHECK
PROPER
OPERATION.

REAR COCKPIT

NEW
NEW
NEW
NEW
NEW

- 1. Boost pumpOFF
- 2. Emergency landing gear retract
switchWIRED
- 3. Emergency fuel switchOFF
- 4. Canopy air pressure2300-3000 PSI
- 5. First Aid kitSECURED
- *6. Lap belt and shoulder harness:
 - a. WebbingCHECK FOR
CUTS
 - b. Release buckleOPERATION
- 7. Parachute/lanyardCONDITION/
CONNECTED
- 8. Inspect for loose gear.
- *9. CanopyCHECK FOR
CRAZING
- *10. Canopy sealSECURITY
- *11. Canopy actuating
mechanismCHECK
PROPER
OPERATION

If Solo Flight:

- 12. Shoulder harness, seat belt, seat cushion, parachute and any loose equipmentSECURED
- 13. Radio extension cordSECURED
- 14. GyroCAGED
- 15. Instrument panel.....SECURED
- 16. CanopyCLOSED AND LOCKED

EXTERIOR INSPECTION

TRAILING EDGE, PORT WING

- 1. Top and underside of wing for cracks, deep scratches, tears, wrinkles, popped rivets, and bulges.
- 2. Movement of aileron, servo action of aileron trim tabs, aileron bellcrank, and trim tab linkage.
- 3. Trim tab hinge pin anchored to hinge pin hole, actuator bolt not cotter-keyed.
- 4. Flaps for obvious damage.
- 5. Static discharge wicks for fraying, deterioration, and proper attachment (min. length 6" overall, 1" exposed).

PORT WINGTIP

Dents, scratches, and condition of navigation light.

LEADING EDGE, PORT WING

- 1. Breaks, bulges, and proper contour.
- 2. Landing light for security of lens and bulb.
- 3. Pitot tube — ensure that pitot tube cover is removed and tube is aligned, secure and unobstructed.

4. Visually check fuel quantity, check chain secure at both ends, and check O ring for deterioration. Replace cap securely.
5. Cockpit air intake screen for obstructions.
6. Undersurface for evidence of fuel leakage.
7. Spar cap for corrosion, adjacent wing skin for bulges.

PORT MAIN LANDING GEAR

1. Condition of main gear doors. Ensure that wheel well is free of obstructions.
2. Uplock bracket spring is attached to uplock bracket and rib of wing.
3. Check small spring inside canvas dust cover attached to uplock cable and the uplock bracket.
4. Uplock mechanism for distortion and security.
5. Roller on landing gear brace for freedom of movement.
6. Shock strut for scoring and pitting; piston for evidence of leakage. Check for approximately 3 inches of polished strut showing.
- *7. Clean exposed area with a clean cloth moistened with hydraulic fluid, leaving a thin film of fluid.
8. Hydraulic fittings for condition and leaks.
9. Brake disc for freedom of movement (parking brake off).
10. Brake puck pin for 3/16" clearance maximum.
11. Check wheel nut dust cap free to turn.
12. Tire for condition and proper inflation.

FUEL SAMPLES

1. Open port engine compartment cowl.
2. Front cockpit fuel shutoff valve handle to ON.
3. Boost pump switch to ON.
4. Battery switch to ON.
5. Take fuel sample from AFC 53 drain.
6. Open access panel 4, take fuel sample from fuel sump and close access panel.
7. Open access panel 7, take fuel sample from main fuel strainer drain and close access panel.
8. Battery switch, boost pump, and fuel shutoff valve handle to OFF.

PORT ENGINE COMPARTMENT

1. Apron beneath engine nacelle for excessive gasoline, oil, or fluid leakage. Underside of fuselage for excessive oil leakage.
2. Fuel filter if AFC 53 installed. Fuel Drain switch in closed position, red plunger flush with top of filter assembly, and prop governor control line not chafing on filter mounting bracket.

WARNING

If the fuel drain switch is not in the closed position, fuel will be pumped overboard when the battery switch is turned on. Fuel discharge will continue in flight, cause rapid depletion and could result in fuel starvation within 20 minutes.

NEW
NEW
NEW
NEW
NEWNEW
NEW

3. Oil level (minimum of 10 quarts for cold engine; 11 quarts for hot engine).

WARNING

Before re-installing oil filler cap, ensure retaining chain is not broken and is attached at both ends.

4. Engine for loose fittings and leaks.
5. Alternate air door for security.
6. Obstructions in augmentor tube.
7. Ensure that cowling is latched after engine inspection.

NOSE SECTION

1. Wheel centering mechanism, nose gear bracket and retract arm for looseness.
2. Nose gear centering roller free to move.
3. Uplock mechanism for distortion and security.
- *4. Nose gear shimmy damper for evidence of leakage and 1/16 to 1/32 inch of polished rod showing when nose wheel is turned to stops in both directions.
5. Shock strut scoring, pitting or leakage and approximately 5 inches of extension.
- *6. Clean exposed area of strut with clean cloth moistened with hydraulic fluid, then wipe clean leaving a thin film of hydraulic fluid.
7. Mud scraper for security and obvious damage
8. Grounding wire for security and touching the deck.
9. Nose tire for condition and proper inflation.
10. Propeller for cracks, nicks, pits, and evidence of oil leakage. Inspect security of spinner retention nut.

- N
E
W**
11. Airscoop screen and hot air overboard inlets for cleanliness and obstructions. Lock fasteners firmly secured.
 12. Check passing light for security.

STARBOARD ENGINE SECTION

- N
E
W**
1. Same as check for port engine section except for oil quantity. Check for double clamp on generator capacitor.
 2. Security of external canopy release handle.
 - *3. Battery for electrolyte leakage, warps and bulges.
 - *4. Battery and connectors for security.
 - *5. Vent tubes for obstructions. Ensure vent tube is properly connected to vent elbow.
 - *6. Drain jar for evidence of contamination.

STARBOARD MAIN LANDING GEAR

Same as port main landing gear.

STARBOARD WING

Same as port wing.

FUSELAGE, STARBOARD SIDE

- N
E
W**
1. Fuel vent standpipe for dents and for a forward slant of approximately 15 degrees with bias cut facing forward.
 2. Wrinkled skin and popped rivets.
 3. VOR antenna for security.
 4. Static air vent clear.

EMPENNAGE

1. All tail surfaces for cracks, corrosion, dents and tears; control surfaces for freedom of movement, looseness and excessive play between elevators.
2. All visible control linkages and hinge fittings for cracks and security. Rudder trim tab for anti-servo action.
- *3. Retaining nut on VHF antenna for security. Antenna for cracks, dents, corrosion and security.
- *4. Navigation lights and tail cone for security and cracks.
- *5. Static discharge wicks for fraying, deterioration and proper attachment. (Min. length 6", 1" exposed wick).

FUSELAGE, PORT SIDE

Same as starboard side except for ensuring that baggage compartment is checked and securely latched.

WARNING

The baggage compartment must be checked empty for dual flights. A maximum of 100 pounds may be carried in the baggage compartment on solo flights.

AIRCRAFT EXTERIOR

At unmanned fields or fields where ground support personnel are not normally available, the pilot will ensure that the aircraft wheel chocks and tiedowns are removed and the location of the nearest fire bottle is noted.

PRESTART**PRESTART CHECKLIST**

1. Seat and rudder pedalsADJUSTED
2. HarnessFASTENED
3. Inertia reel lock.....CHECKED
4. Wing flap lever.....OFF
5. Landing lightsOFF
6. Alternate airIN
7. Inverter.....OFF
8. GeneratorON
9. Cockpit air handlesSET
10. Landing gear emergency
handcrankDISENGAGED
(clutch knob Up
and LOCKED)
11. Light switches and rheostatsSET
(The anti-collision switches will normally
remain in the ON position at all times).
12. Radios.....OFF
13. Pitot heatOFF
14. Circuit breakers.....IN

On night flights only:

15. External power (for lights
and gyros).....PLUGGED IN
(if not available,
battery switch
ON)
16. Instrument and console lights ...ON
17. Landing lights, passing light,
and pitot heatCHECK (ON
momentarily)

- 18. Navigation lightsON (checked by outside observer)
- 19. External gear-down indicator lights.....CHECKED (by outside observer)

STARTING ENGINE (FRONT COCKPIT ONLY)

N I W

- 1. CanopyOPEN
- 2. ~~Mixture~~.....~~IDLE CUTOFF~~
- 3. Fuel shutoff valve handleON
- 4. Fuel boost pump.....ON
- 5. ~~Throttle~~ ^{MIXTURE FULL}.....~~SET~~
- 6. Propeller control.....FULL FORWARD
- 7. ~~Parking brake~~ ^{MAGNETO BOTH}.....~~SET~~
- 8. BatteryON (OFF if external power is used)
- 9. Fuel pressure.....CHECK
- 10. Propeller areaCLEAR
- 11. ~~Starter~~ ^{THROTTLE FULL (5510), OFF}.....~~ENGAGED (COUNT 4 - BLADES)~~
- 12. ~~Ignition~~.....~~BOTH~~
- 13. Mixture.....SLOWLY TO RICH ←
- 14. Starter.....RELEASE (when engine fires)
- 15. Throttle.....1200-1400 RPM
- 16. Oil pressure.....CHECK (if no rise within 10 seconds or 30 psi in 30 seconds, secure engine)

WARM ENGINE

OFF

1/4" FORWARD
1/2" FORWARD

SLOWLY TO RICH ←

- 17. External power (if used)DISCONNECT;
Battery - ON
- 18. Radios/AIMS.ON/STBY

If engine fails to start after ~~15~~¹⁰ seconds: (*Flooded Engine*)

- 19. MixtureIDLE CUTOFF
- 20. ThrottleFULL OPEN
- ~~21. IgnitionOFF~~
- 22. Crank for ~~5~~¹⁰ seconds to clear engine, then repeat steps ~~5 through 14~~. If engine fails to start after 10 seconds, perform steps ~~23 through 29~~. *COOL 5 MIN*
- 23. MixtureIDLE CUTOFF
- 24. Booster pumpOFF
- 25. BatteryOFF
- 26. IgnitionOFF
- 27. Fuel shutoff valve handleOFF
- 28. Allow starter to cool for 5 minutes, then repeat steps 2 through 14.
- 29. If engine still fails to start, secure engine and request assistance.

PRETAKEOFF

PRETAXI CHECKLIST

- 1. Generator warning light and generator voltageOUT AT CUT-IN SPEED
(approximately 900 engine rpm), 27.7-28.5 VOLTS
- 2. FlapsCYCLED/INDICATING UP
- 3. Trim tabsSET 6R, 3 UP, 0
- 4. Landing gear warning lightCHECKED

5. Landing gear indicatorsDOWN
6. Fuel quantityNOTED
7. Altimeter and clockSET
8. InvertersCHECKED/ON
MAIN
9. GyrosUNCAGED/SET
10. Communications equipmentCHECK
OPERATION
11. InstrumentsCHECK FOR
CORRECT INDI-
CATIONS
12. Idle speedTHROTTLE
CLOSED, 600-
750 RPM
13. Ignition groundCHECKED

TAXI

1. Parking brakeRELEASE
2. BrakesCHECK
3. Throttle to taxi speed800-1000 RPM
4. Turn-and-slip indicator and
directional gyroCHECK FOR
PROPER
TRACKING

ENGINE RUNUP

1. InstrumentsCHECK FOR
CORRECT INDI-
CATIONS
2. Propeller(recheck)FULL
INCREASE
3. Mixture (recheck)FULL RICH

Propeller Governor Check:

- 4. Throttle.1800 RPM
- 5. Propeller leverAFT TO
KEEP ABOVE 1500 RPMs DETENT (drop
to 1600-1650
rpm)
- 6. Propeller leverRETURN TO
FULL
INCREASE

Exercise propeller by repeating this procedure.

Ignition System Check:

- 7. Throttle.2000 PRM
- 8. Ignition switchR (Right). (Note
rpm drop, then
switch to BOTH)
- 9. Ignition switchL (Left). (Note
rpm drop, then
switch to BOTH.
Maximum drop
on either mag-
neto - 100 RPM
with a max. split
of 50 RPM)

Alternate Air Check:

- 10. Throttle.2000 RPM
- 11. Alternate air handle.FULL OUT
(manifold pres-
sure drop -
approx. 1/2 inch)
- 12. Alternate air handle.FULL IN

- 13. Throttle. FULL OPEN
(2475(±75)rpm)
- 14. Acceleration and deceleration should be smooth
without backfire or roughness.
- 15. Fuel boost pump. OFF AT 1700
RPM; CHECK
PRESSURE
FLUCTUATION
- 16. Fuel boost pump. ON; CHECK
15-20 PSI
- 17. Idle mixture:
 - Throttle. CLOSED (600 —
750 rpm)
 - Mixture control. SLOWLY RE-
TARD TO IDLE
CUTOFF (5-10
rpm rise)
 - Mixture control. RAPIDLY TO
FULL RICH (as
soon as rpm
starts to fall off)

GROUND BURNOUT

If, after extended taxi or idle time, the ignition system checks out of limits, proceed as follows:

- 1. Propeller FULL
INCREASE
- 2. Mixture FULL RICH
- 3. Throttle. ADVANCE TO
2,000 RPM
- 4. Mixture LEAN TO 50
RPM DROP
BELOW BEST
POWER

- | | |
|----------------------------------------|-------------------------------|
| 5. Mixture | RETURN TO RICH AFTER 1 MINUTE |
| 6. Ignition system | RECHECK |
| 7. If system does not check out. . . . | DOWN THE AIRCRAFT |

TAKEOFF

TAKEOFF CHECKLIST

- | | |
|---------------------------------|----------------------|
| 1. Anti-collision lights | ON |
| 2. Fuel. | ON; CHECK: |
| Fuel boost pump | ON |
| Fuel shutoff valve handle | ON |
| Fuel pressure | NORMAL |
| Fuel quantity | CHECK |
| Fuel caps | LOCKED |
| 3. Flight controls | FREE |
| 4. Trim tabs | 6 R, 3 UP, 0 |
| 5. Alternate air | OFF |
| 6. Generator | OPERATING |
| 7. Instruments | CHECKED/SET |
| 8. Mixture | RICH |
| 9. Propeller | FULL INCREASE |
| 10. Ignition | CHECKED ON BOTH |
| 11. Flaps | SET |
| 12. Harness | LOCKED |
| 13. Canopy: | |
| Front | LOCKED (Open/closed) |

- Rear LOCKED
(Closed)
- 14. Radios SET
- 15. AIMS ON

AFTER TAKEOFF CHECKLIST

- 1. Landing gear UP
- 2. Flaps UP
- 3. Fuel caps SECURE
(VISUALLY
CHECK)

LANDING

LANDING CHECKLIST:

- 1. Fuel boost pump ON
- 2. Alt air IN
- 3. Mixture RICH
- 4. Harness LOCKED
- 5. Landing gear DOWN
- 6. Flaps AS REQUIRED
- 7. Propeller FULL
INCREASE

SECURE CHECKLIST

- 1. Throttle 800-1000 RPM
- 2. Flaps UP
- 3. Radios OFF
- 4. Inverters OFF
- 5. Engine instruments WITHIN LIMITS/
TEMPS STABI-
LIZED
- 6. Throttle CLOSED
- 7. Ignition GROUND-
CHECK
- 8. Fuel boost pump OFF
- 9. Mixture IDLE CUTOFF

After propeller stops:

10. Fuel shutoff handleOFF
11. IgnitionOFF
12. LightsOFF
13. BatteryOFF

N-1

Before Leaving Aircraft:

1. Flight controlsLOCKED
2. WheelsCHOCKED
3. Parking brakeRELEASED
4. CanopyCLOSED

SPECIAL PROCEDURES

TABLE OF CONTENTS

	Page
ACROBATICS	47
STALL RECOVERY	48
SPIN RECOVERY	49

ACROBATICS

ACROBATIC CHECKLIST

Complete the following prior to performing acrobatics:

1. Fuel boost pump ON
2. Harness TIGHT AND LOCKED
3. Propeller 2,400 RPM
4. Canopy CLOSED AND LOCKED
5. Directional Indicator CAGED
6. Loose gear STOWED

NOTE

Complete all acrobatic maneuver recoveries above 3000 feet AGL.

ACROBATIC MANEUVERS

	ENTRY AIRSPEED	RECOVERY AIRSPEED
WINGOVER	130 knots IAS. Air speed should be 70 knots IAS after completing 90° of turn.	Recover 180° from original heading with 130 knots IAS.
BARREL ROLL	130 knots IAS. When inverted, and after 90° heading change, airspeed should be 60-70 knots.	Recover on original heading with 130 knots IAS.
LOOP	150 knots IAS.	150 knots IAS.
HALF CUBAN EIGHT	150 knots IAS.	Recover 180° from original heading at 150 knots IAS.
IMMELMANN	170 knots IAS.	
SPLIT-S	90 knots or less. When inverted, wings level, close throttle.	Recover on opposite heading, 140 knots or less.

STALL RECOVERY

NORMAL STALL RECOVERY

1. Release back pressure on stick immediately and smoothly advance throttle.
2. Roll wings level and return to level flight. Avoid pulling back too severely as a secondary stall or excessive G-loads may result.
3. When level flight is resumed, reduce throttle to cruising power.

LOW ALTITUDE STALL RECOVERY

1. Use power to hold altitude loss to a minimum. The nose of the aircraft should be allowed to drop only slightly below the horizon.
2. Use coordinated aileron and rudder to roll the wings level and return to level flight as rapidly as possible.
3. As soon as control is regained, establish climb.

SPIN RECOVERY

Recovery from normal spins is effected most rapidly if started at the beginning of the steep half of the turn. Recovery is equally positive in the shallow portion, but is somewhat slower.

1. Apply opposite rudder to the neutral position followed by forward stick to the neutral position.
2. When the rotation stops; level the wings. The aircraft will be in a 60-to 80-degree dive. Start a pullout immediately to keep the altitude loss to a minimum, but avoid entering an accelerated stall.
3. With gear and flaps down, make pullout tight enough to keep from exceeding 100 knots IAS.

REFERENCE DATA

TABLE OF CONTENTS

	Page
LIMITATIONS.....	51
SERVICING	52

LIMITATIONS

AIRSPPEED

- In smooth or moderately turbulent air:
 Gear and flaps retracted,
 canopy open or closed 240 KIAS (max.)
- In severe turbulence 120 to 165 KIAS

ACCELERATION

- In smooth air at all gross weights +4.0 to -2.0 G's
- In moderately turbulent air at all
 gross weights +2.5 G's

ENGINE

- Minimum speed 1600 RPM
- Maximum speed 2600 RPM
- Engine overspeeds (2700-3200 rpm)
 require engine inspection; engine
 speeds above 3200 rpm require
 engine change.

Cylinder head temperature:
 Normal operating range 107° C to 240° C
 Maximum continuous 240° C
 Takeoff 240° C

Fuel pressure:
 Continuous operation 15-20 PSI

Oil pressure:
 Normal operating range..... 30 to 80 PSI
 Maximum 80 PSI

Oil temperature:
 Normal operating range 40° C to 107° C
 Maximum 107° C

PROPELLER

Overspeed (maximum) 3380 RPM
 Propeller inspection is required
 between 3050 - 3380 rpm; replacement
 is required over 3380 rpm.

GROSS WEIGHT

Field takeoff and landing 3050 LB. (max.)

SERVICING

ENGINE STARTING UNITS

External (auxiliary) power units may be used which supply
 24 volts dc.

OLEO EXTENTIONS

Main landing gear strut – Extended 3 inches (minimum).
 Nose gear strut – 3-3/16 to 5-3/16 inches.
 Nose gear shimmy damper – Service shimmy damper if indi-
 cator wire can be inserted into shimmy damper piston rod
 3-1/16 inches or more.

LIMITATIONS CARD

	MIN	MAX
MAP SPLIT		1.5"
OIL TEMP (normal range)	40°C	107°C
OIL PRESSURE (in flight)	30 psi	80 psi
OIL PRESSURE (idle)	10 psi	
OIL CONSUMPTION (1.0 hour normal cruise)		1.5 g/s
CYL. HEAD TEMP (normal range)	107°C	240°C
CNT SPLIT		30°C
FUEL PRESSURE (start) 8 psi (normal)		15-20 psi
FUEL SPLIT (must be dipped)		10 gals
RPM SPLIT		20 rpm
RPM FLUX		±25 rpm
RPM (full power on deck)	2400 rpm	2550 rpm
RPM (takeoff)	2570 rpm	2630 rpm
RPM (idle at operating temp)	800 rpm	750 rpm
VOLTMETER (at 1700 rpm)	27.7 V	28.5 V
ALTIMETER SPLIT (below 1000')		100 ft
AIRSPD SPLIT		4 kts
STALL SPEED (gear, flaps down, power off)		49.3 kts
STALL SPEED (clean, power off)		59.3 kts
GEAR WARNING HORN BLOWING	12"	18"
LANDING GEAR RUN TIME (retract)	7-8 sec	12 sec
FLAP RUN TIME (extension)		15 sec
BRAKE PUCK WEAR		3/16"
AILERON TRIM (normal cruise)		4°
IDLE MIXTURE CHECK		5-10 rpm rise
CRUISE MIXTURE RISE		0-40 rpm rise
MAGNETO DROP		100 rpm
MAGNETO SPLIT		50 rpm
MAXIMUM AIRSPD (clean)		240 kts
ROLLING PULLOUT		2.5 g.s
GEAR DOWN		110 kts
"O" METER LIMITATION		+4, -2
CANOPY AIR	2500 psi	3000 psi

NOTE: IF A READING IS BELOW THE MINIMUM OR ABOVE THE MAXIMUM, THE AIRCRAFT IS DOWN ON THE YELLOW SHEET.

NAVAIR 01-90KDB-1B

BATTERY

Capacity: 24 volt, 24 ampere-hour.

Specific gravity, fully charged: 1.275 to 1.300 (recharge or replace when specific gravity is 1.240 or below).

SPECIFICATIONS

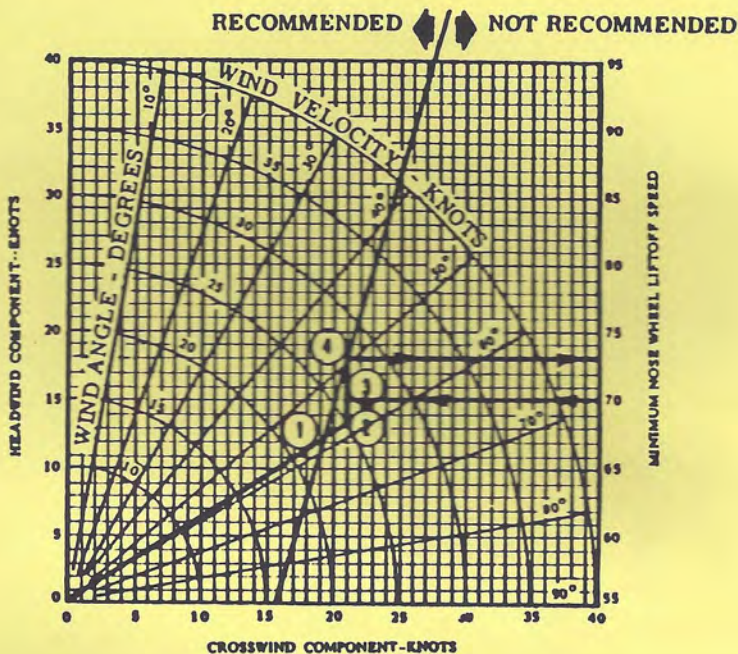
MATERIAL	SPECIFICATIONS	QUANTITY OR PRESSURE
FUEL	MIL-F-5572** Grade 80/87, 100LL Alternate Grades 100/130, 115/145	50 Gallons (usable fuel, both tanks)
ENGINE OIL	MIL-L-22851 (Type II)**	3 Gallons
HYDRAULIC FLUID	MIL-H-83282 (Primary) MIL-H-56106 Alternate**	1 Pint Fill brake reservoir to within 3/4 inch of bottom of reservoir neck.
LANDING GEAR STRUTS Hydraulic Fluid Nitrogen*	MIL-H-5606** BB-N-411	_____
TIRES Dry Compressed Air or Nitrogen* Main (6.50 X 8-6-ply) Nose (5 X 5-4-ply)	_____	35 psi 40 psi
CANOPY EMERGENCY AIR BOTTLE Nitrogen*	BB-N-411	2,300 - 3,000 psi
BATTERY Distilled Water	_____	Electrolyte level must cover plates in each cell.

*If necessary, on a cross country flight, air may be used on a one time basis. Since air is corrosive, the system shall be returned to nitrogen at the next stop or home base as applicable. An appropriate yellow sheet entry shall be made when air is substituted.

**Commercial - Fuel: 80/87, 100LL, alt: 100/130 or 115/145; Oil: EXXON Aviation Oil E120 or E100, EXXON Aviation Oil AD100, Aeroshell W120 or W100, Mobil AVREX 106 type 120, Mobil AERO Oil 120 or 100, Chevron AERO Oil Grade 120 or 100, Texaco Aircraft Eng. Oil Premium AD120 or AD100 (Lower viscosity aviation grade oils may be utilized during cold weather operations); Hydraulic Fluid: EXXON UNIVIS J43, Mobil AERO HFB, Aeroshell Fluid 4.

TAKEOFF CROSSWIND CHART

MODEL T-34B

ENGINE (1)
NO.0-470-4

SAMPLE PROBLEM:

KNOWN:
TAKEOFF RUNWAY - 2
WIND - 078°/25 KNOTS

DETERMINE:
IF TAKEOFF IS RECOMMENDED AT
TAKEOFF SPEED OF 70 KNOTS IAS.

NOTE

- MAXIMUM NOSEWHEEL LIFT OFF SPEED IS 90 KNOTS IAS.
- ENTER CHART AT MAXIMUM GUST VELOCITY.

DATA BASED ON: FLIGHT TEST

SOLUTION:

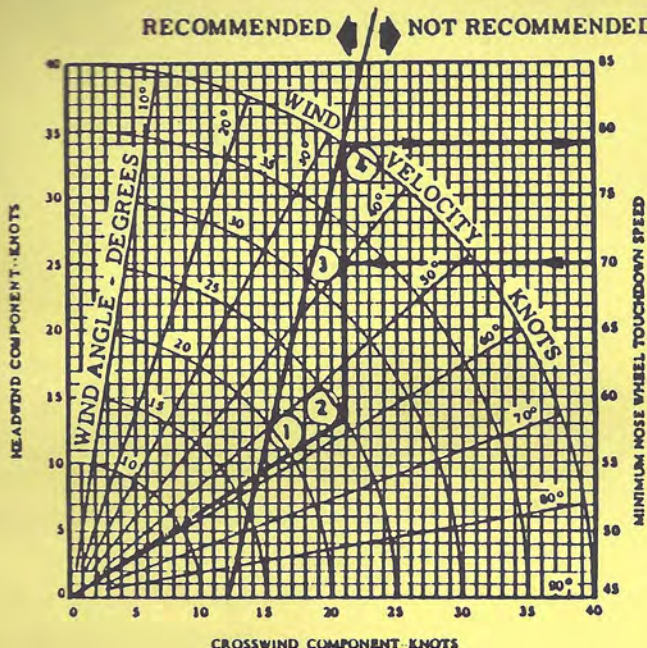
- ① WIND ANGLE IS $78^\circ - 20^\circ = 58^\circ$ WIND ANGLE TO RUNWAY HEADING.
- ② AT WIND VELOCITY OF 25 KNOTS AND 58° WIND ANGLE TO RUNWAY HEADING. FIND CROSSWIND COMPONENT OF 21 KNOTS AND HEADWIND COMPONENT OF 13 KNOTS.
- ③ PROCEED VERTICALLY TO PREDICTED TAKEOFF SPEED OF 70 KNOTS IAS AND DETERMINE TAKEOFF AS NOT RECOMMENDED.
- ④ CONTINUE VERTICALLY AND DETERMINE THAT TAKEOFF SPEED MUST BE 73 KNOTS IAS BEFORE TAKEOFF IS RECOMMENDED.

FUEL GRADE: 80/87
FUEL DENSITY: 6 LBGAL

LANDING CROSSWIND CHART

MODEL T-34B

ENGINE (1)
NO.0-470-4



SAMPLE PROBLEM.

KNOWN
LANDING RUNWAY - 2
WIND - 070° 25 KNOTS

DETERMINE
IF LANDING IS RECOMMENDED AT
LANDING SPEED OF 70 KNOTS IAS.

NOTE

- MAXIMUM NOSE WHEEL TOUCHDOWN SPEED IS 90 KNOTS IAS.
- ENTER CHART AT MAXIMUM GUST VELOCITY.

SOLUTION:

- ① WIND ANGLE IS 70° - 20° - 50° WIND ANGLE TO RUNWAY HEADING.
- ② AT WIND VELOCITY OF 25 KNOTS AND 50° WIND ANGLE TO RUNWAY HEADING, FIND CROSSWIND COMPONENT OF 21 KNOTS AND HEADWIND COMPONENT OF 13 KNOTS.
- ③ PROCEED VERTICALLY TO PREDICTED LANDING SPEED OF 70 KNOTS IAS AND DETERMINE LANDING AS NOT RECOMMENDED.
- ④ CONTINUE VERTICALLY AND DETERMINE THAT LANDING SPEED MUST BE 70 KNOTS IAS BEFORE LANDING IS RECOMMENDED.

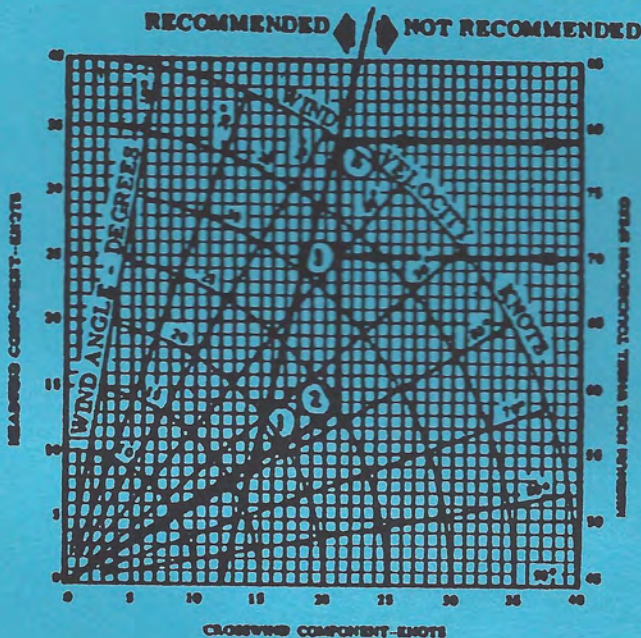
ORIGINATOR: PERMUT 1153

FIGURE 80-7
FIGURE 810-1A1

LANDING CROSSWIND CHART

MODEL T-34B

ENGINE (1)
NO.0-470-4



SAMPLE PROBLEM:

KNOWN:
LANDING RUNWAY - 3
WIND - 070° 25 KNOTS

TO DETERMINE:
IF LANDING IS RECOMMENDED AT
LANDING SPEED OF 70 KNOTS IAS.

NOTE

- MAXIMUM IAS FOR T-34B IS 90 KNOTS IAS.
- ENTER CHART AT MAXIMUM GUST VELOCITY.

SOLUTION:

- ① WIND ANGLE IS 70° - 20° 50° WIND ANGLE TO RUNWAY HEADING.
- ② AT WIND VELOCITY OF 25 KNOTS AND 50° WIND ANGLE TO RUNWAY HEADING, FIND CROSSWIND COMPONENT OF 21 KNOTS AND HEADWIND COMPONENT OF 13 KNOTS.
- ③ PROCEED VERTICALLY TO DETERMINE LANDING SPEED OF 70 KNOTS IAS AND DETERMINE LANDING AS NOT RECOMMENDED.
- ④ CONTINUE VERTICALLY AND DETERMINE THAT LANDING SPEED MUST BE 70 KNOTS IAS BEFORE LANDING IS RECOMMENDED.

DATE: 10-10-66 (REVISED)

FIGURE NO. 1
FIGURE NO. 101-10