PLASTIC MOVELTY ITEMS

An Infrequent Newsletter From IPMS / CHICAGO SPRUE STRETCHERS

BUZZ OFF

by Jim Griffiths

Background

In June, 1944, a week after the Normandy landings, the Germans attacked England with a new weapon—a flying bomb. The Germans had hoped to bring this weapon into use during the previous autumn with the intent of bombarding London with upwards of as many as 3000 missiles a week until the destruction of the city had reached acceptable proportions. While the timetable had been thrown off, the intention remained: flying bombs would rise from secret installations built in a great arc throughout north—ern France to rain down on the enemy's cities.

In attacking large targets by remote means, the V-1 was unrivaled for economy and simplicity, despite the great number of modifications. More than 50 V-1 bombs could be had for the cost of a single V-2 rocket, and more than 300 V-1's for the expense of a single bomber. Moreover, each flying bomb delivered as much explosive as a V-2 or a bomber.

While inexpensive, simple and sufficient for the range, the V-1's pulsejet had shortcomings: low altitude and insufficient speed. Also a fault in the fuel-starved pulsejet caused it to cut out during the bomb's dive. This act provided warning to those below and also minimized penetration.

Although Americans called the V-1 by the name "Buzz Bomb", the English called it the "Doodlebug" (originated in No. 486 Tempest Squadron) or Chuff-Bomb" (originated by No. 69 Mosquito Squad.). The Germans called it V-1("Vergeltungswaffe Eins": Retaliation Weapon No. 1), though sharp-witted Berliners were thought to have deduced another explanation of V-1: Versager Eins(Flop No. 1).

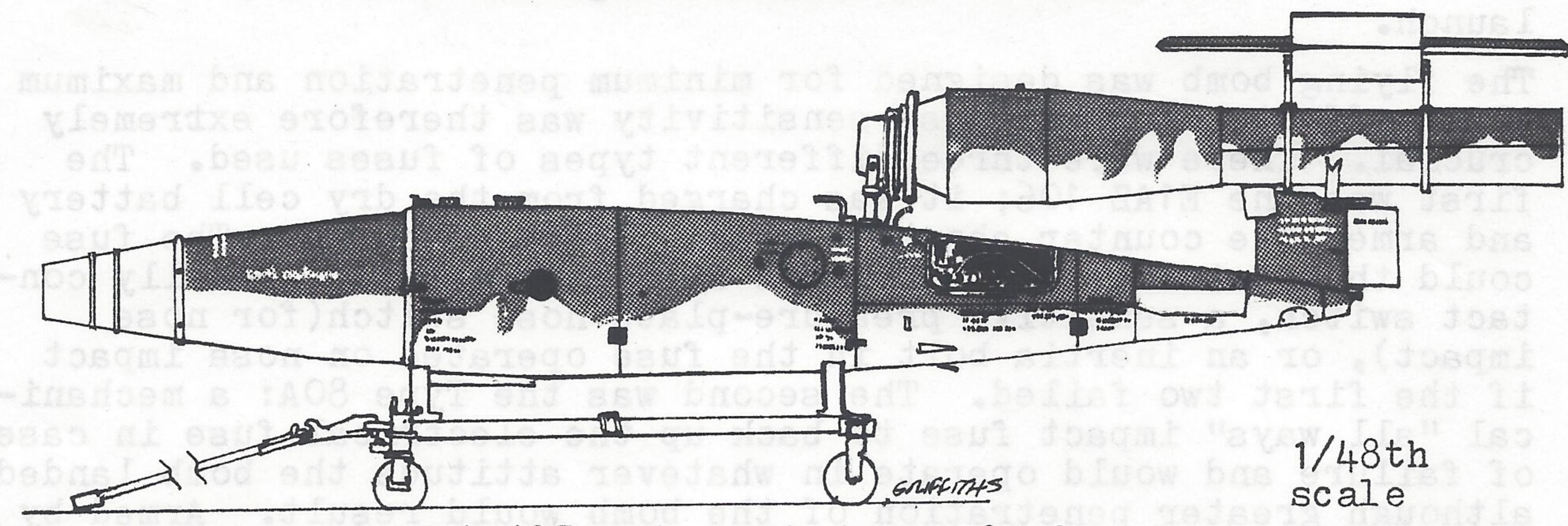
The Fi-103 was the first operational weapon to be driven by y form of jet propulsion. It was a pilotless mid-wing monopl without ailerons, but it incorporated a conventional tailplane with elevators, set forward from a fin and rudder. The propulsion and guidance systems depended on a continuous supply of two sources of power:compressed air and electricity. The air was stored in two spherical air bottles; the electricity was provided by a 42-cell electric dry battery.

The V-1 could be launched in two ways: from a ground-based ramp or from an aircraft. In the former, the bomb climbed at an angle equal to that of the launching ramp until it reached a pre-determined altitude when it levelled off and turned on a pre-determined heading. It then flew straight and level until it had covered the required distance as recorded by the air-log on a mechanical counter. At this point, the bomb dived.

Air launching of V-1's from Heinkel He-111 "carrier" or "parent" bombers was, for the most part, carried out against London. But no matter what the intended target, the big problem for the attackers was to put the "carrier" into the correct position and on the correct bearing to release the bomb at a precisely calculated moment. A semi-automatic timing devise installed

in the "carrier" ensured the highest degree of accuracy at launch.

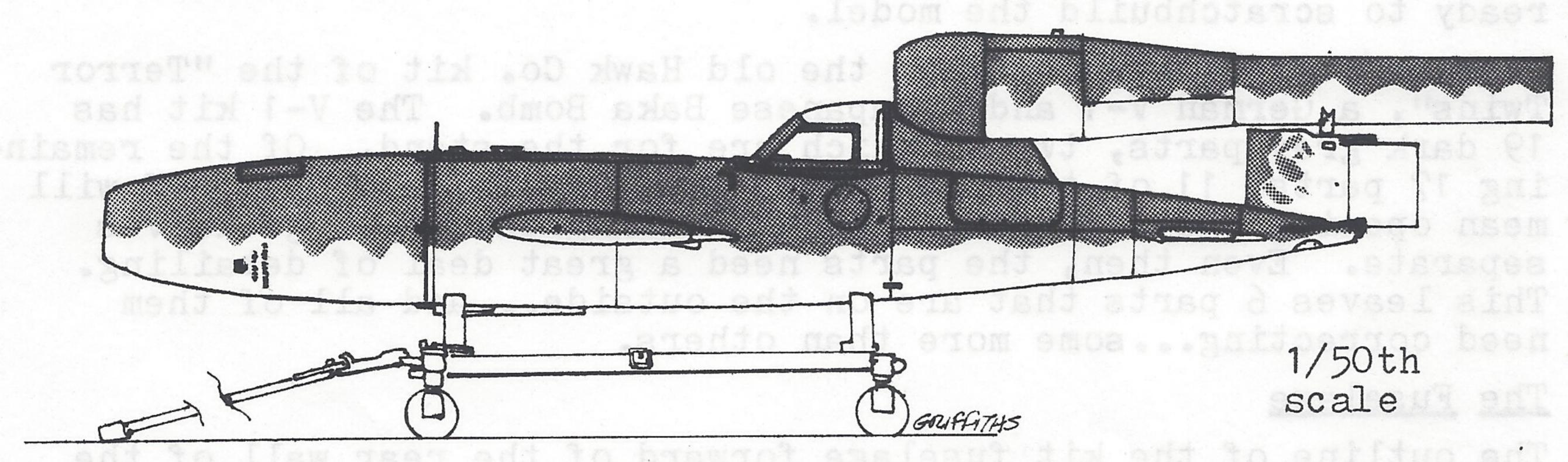
The flying bomb was designed for minimum penetration and maximum blast effect. Warhead fuse sensitivity was therefore extremely crucial. There were three different types of fuses used. The first was the E1AZ 106; it was charged from the dry cell battery and armed the counter about 38 miles after launching. The fuse could then detonate by one of the following methods: a belly contact switch, a sensitive pressure-plate nose switch (for nose impact), or an inertia bolt in the fuse operated on nose impact if the first two failed. The second was the Type 80A: a mechanical "all ways" impact fuse to back up the electrical fuse in case of failure and would operate in whatever attitude the bomb landed, although greater penetration of the bomb would result. Armed by a clockwork mechanism within the fuse, it was started by the removal of a pin on launching. It was armed within 8 minutes(max.) following the launch. The third and last fuse was the Type 17B. It was armed electrically on launching, following the withdrawal of a pin causing a clockwork delay mechanism to start. Max. Delay was 2 hours from point of arming.



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Fi-103 A-1, werk no. 478 180, Dannenberg, Germany, April 25, 1945

When advancing troops of the U.S. Army's 29th Inf. Div. took the vast complex of Dannenberg, this V-1 was found. Appearing to have Color 82 uppersurfaces and Color 76 undersurfaces, an unusual feature is the camouflaging of the inner warhead in 82/76. A shipping container covers the electrical impact fuses. The horizontal stabilizer in its shipping container attached to the pulsejet exhaust tube with wire bands; the container itself is made of wood. This V-1 has the pulsejet intake cover and pylon fairing missing, as well as the access panel to fuel pump, filter, & counter mechanism. Another unusual feature is the coloring of the fin and rudder: the fin appears in 82 with the rudder in 76. Note how camouflage pattern varies from section to section: this is due to prefabrication & painting of parts in different locations w/ assembly at a later time.



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Fi-103 A-1/Re 4, wk. no. unknown Dannenberg, Germany, April 25, 1945

I opted for method (5).

This piloted version of the V-1 differed from the A-1 version in that its length was longer, the wingspan & chord were longer, the rudder was longer(w/ control linkages on p&s sides), the warhead cover was made of plywood, the forward pylon fairing's shape was different, and of course, it had a cockpit & canopy. Colors appear to be 81 uppersurfaces w/ 76 undersurfaces; there appears to be a slight mottling on the fin. V-1's surface is shiny, probably from a coating of wax. Note that the impact fuse hasn't been attached to the plywood nose cone.

PNI Foldout No. 4

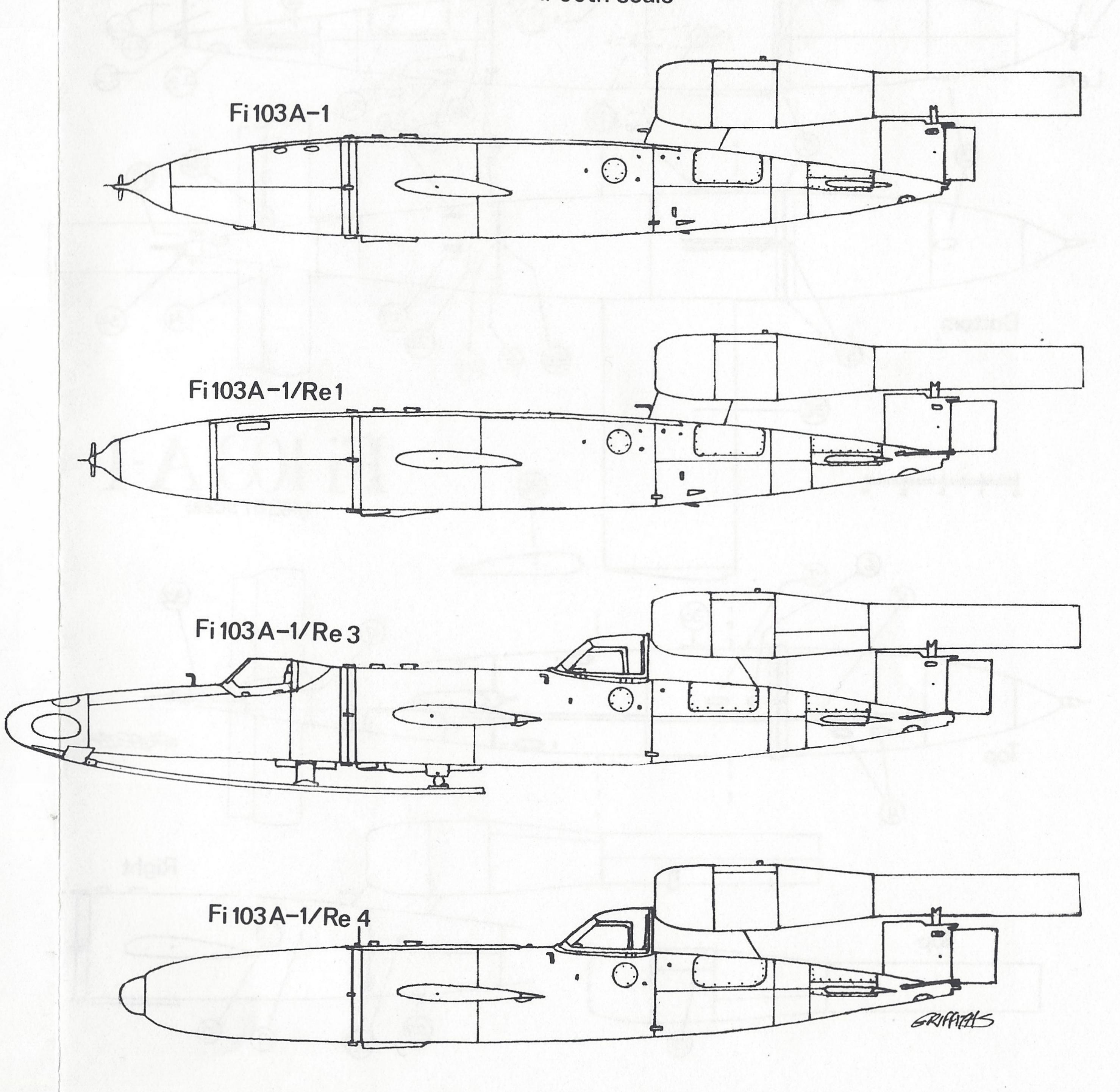
WEIGHTS AND PERFORMANCE STATISTICS *

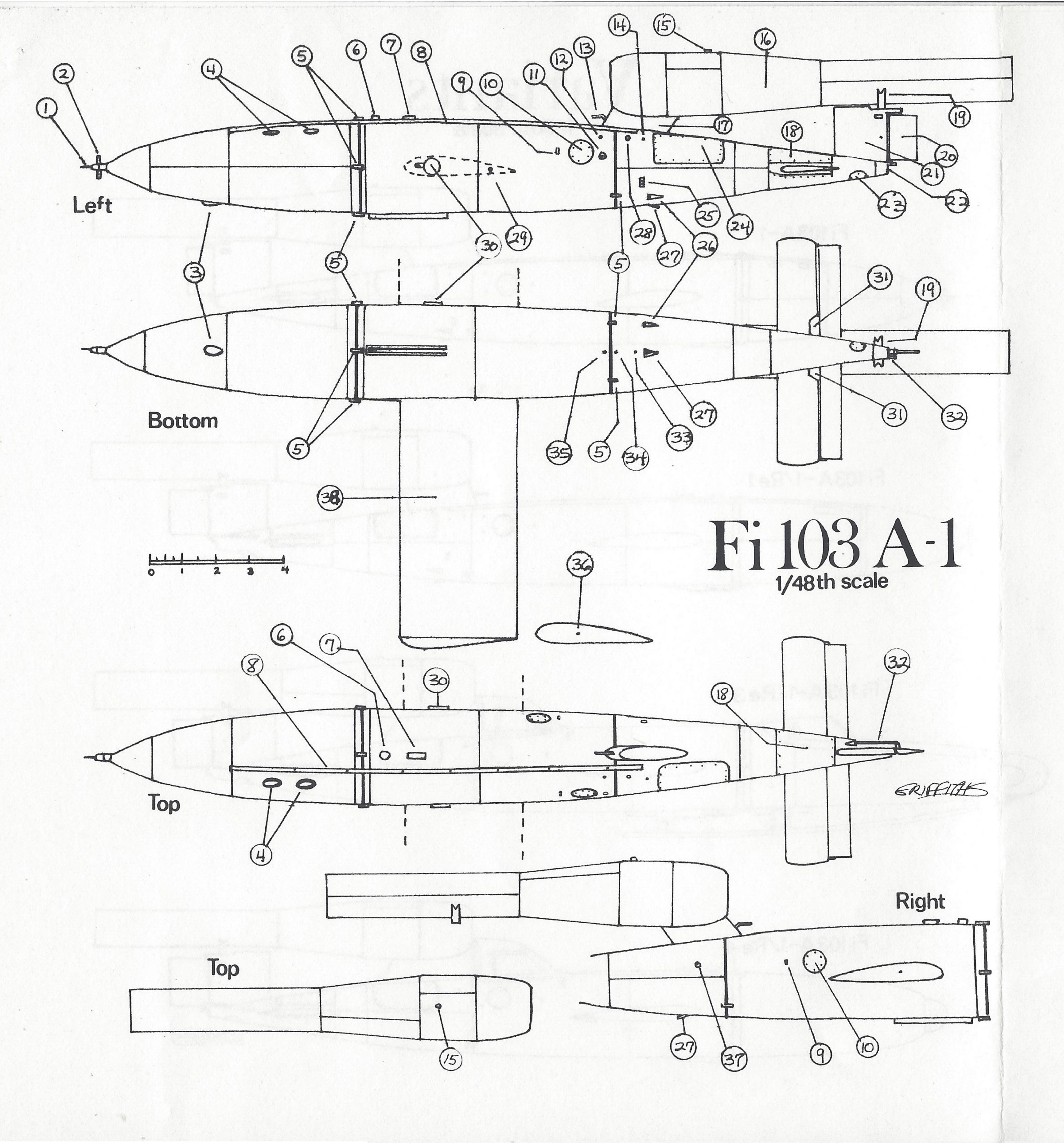
Туре		Fi 103 A-1	Fi 103 A-1/Re 1	Fi 103 A-1/Re 3	Fi 103 A-1/Re 4
Role Seating		Missile	Extd. Range Missile —	Missile Trainer 2	Piloted Missile
Engine		Argus 109-014	Argus 109-014	Argus 109-014	Argus 109-014
Thrust	km-km/h-kp (lb) km-km/h-kp km-km/h-kp	0-0-366 (807) 0-600-325 (716) 3-600-254 (560)	0-0-366 (807) 0-600-325 (716) 3-600-254 (560)	0-0-366 (807) 0-600-325 (716) 3-600-254 (560)	0-0-366 (807) 0-600-325 (716) 3-600-254 (560)
Fuel	Liter (US gal.)	665(176)75 octane	680 (180) VK2	680 (180)	680 (180)
Engine weight Empty weight Fuel weight	kg (lb) kg (lb) kg (lb)	153 (337) 839 (1,864) 500 (1,102)	153 (337) 1,204 (2,675) 513 (1,131)	153 (337)	153 (337)
Crew weight Warhead weight Equipped weight	kg (lb) kg (lb) kg (lb)	830 (1,830) 2,152 (4,796)	450 (1,000) 2,163 (4,806)	202 (448)	101 (224) 810 (1,800)
Fuel consumption Optimum range Service ceiling Optimum cruise speed Maximum speed Rate of climb	L-kp (gal-lbst) km (mi) m (ft) km/h (mph) km/h (mph) m-min (ft-min)	2.59-272 (.68-605) 238 (149) 2.625 (8,840) 580 (360) 644 (400)	2.59-272 (.68-605) 375 (233) 628 (390) 773 (480)	2.59-272 (.68-605)	2.59-272 (.68-605
Speed at launch	km/h (mph)	400 (248)	400 (248)		
Launch time Duration of flight Warhead	sec min	1 25 Amatol-39 (CODE Nr. 52A)	1 40 Amatol-39 (CODE Nr. 52A)		

SPECIFICATIONS **

Туре		Fi 103 A-1	Fi 103 A-1/Re 1	Fi 103 A-1/Re 3	Fi 103 A-1/Re 4
Wing span	mm (ft-in)	5370 (17-61)	5370 (17-61)	5720 (18-9)	5720 (18-9)
Length overall	mm (ft-in)	8325 (27-33)	8509 (27-11)	8929 (29-9)2	8380 (27-2)
Height	mm (ft-in)	1423 (4-8)	1423 (4-8)	1423 (4-8)	1423 (4-8)
Stabilizer span	mm (ft-in)	2055 (6-8 7)	2055 (6-87)	2055 (6-8 7 / ₈)	2055 (6-87)
Fuselage length	mm (ft-in)	7405 (24-3音)	7772 (25-6)	8323 (27-8)2	7780 (25-2)
Maximum fuselage diameter	mm (ft-in)	840 (2-9)	840 (2-9)	840 (2-9)	840 (2-9)
Engine length	mm (ft-in)	3660 (12-01)	3660 (12-01)	3660 (12-01)	3660 (12-01)
Maximum engine diameter	mm (ft-in)	565 (1-101)	565 (1-101)	565 (1-101)	565 (1-101)
Engine orifice diameter	mm (ft-in)	390 (1-33)	390 (1-33)	390 (1-3골)	390 (1-33)
Wing chord	mm (ft-in)	1050 (3-51)	1050 (3-51)	1296 (4-3)	1296 (4-3)
Wing area	m² (ft²)				

Variants 1/50th scale





1. Impact switch to fuses

2. Windmill for range control

3. Belly landing fuse switch

4. Warhead fuse pockets

5. Attachment bolts

6. Fuel tank filler cap

7. Lifting lug

8. Electrical conduit

9. Hoisting eye(one of two)

10. Access plate to compressed air container

11. Air filling valve

12. Air pressure reducing valve

13. Pitot head

14. Push-rod actuating fuel shut-off valve

15. Sparking plug

16. ARGUS As 109-014 pulsejet

17. Pylon fairing cover

18. Access panel to control servo mechanisms

19. Rear pulsejet support band

20. Rudder

21.Fin

22. Hole in rear of fuselage for deployable wire antenna

23. Circular access plate

24. Access plate to fuel pump, filter, and counter mechanism

25. Key operated master switch

26. Starting air quick-release connection

27. Drain from fuel shut-off valve

28. Electrical test socket

29. Swivel housing for wing shear pin

30. Spar housing, locknut and washer

31. Spoilers (NOTE: left underside spoiler is larger than the right one)

32. Rudder control rod

33. Ambient air pressure pipe to altitude control

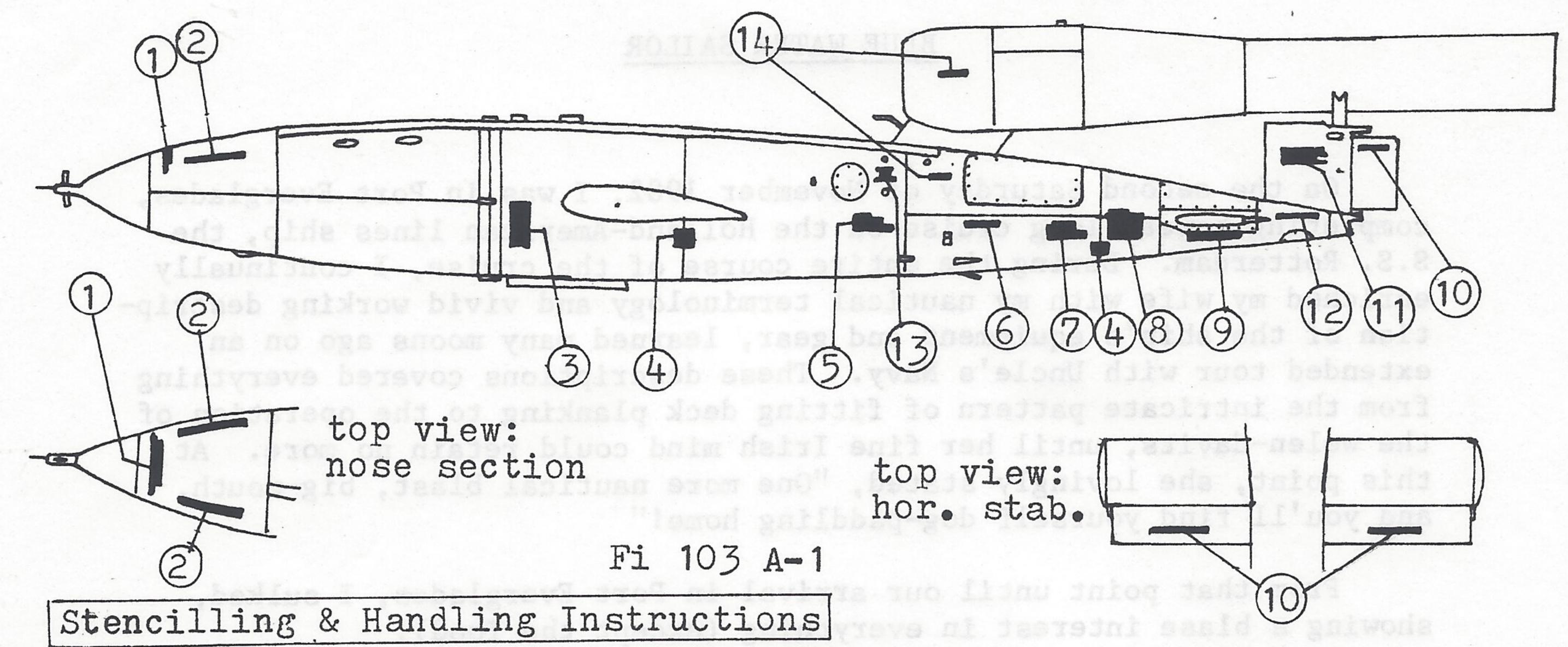
34. Cradle actuated uncaging switch

35. Cradle aligning hole

36. Hole in wingtip

37. Fuel drain(plug)

38. Main wing



- 1. Nasenhaube ist stets mit
- 3. Waggon fz esw bas , "side" gaiwolisd , deck, believing "Holly Shit", and was Zì W 76 N IT 12 November 100 to the back of the bulkheds and 12 November 100 to Rollpallung N Trolley N some stilly regulations involving teartiff, duty and narcottes. WO Abstellpallung A Doppelpallung
- esw sas it sved blbow koul sa 5. TW 76A N Rollpallung lang N Zubringerwagen Conveyor trolley (Schlitten) (Sledge)
- 6. Klebestreifen auf 6. Remove adhesive strip Unterseite vor Inbetriebnahme entfernen
- 7. TW 76 A
- 8. Kfz Verladung 8. Transport vehicle Herkules N Abstellpallung Storage cradle N
- 9. Abstellbock
- 10. Nicht anfassen
- 11. Stutzkeil hier einsetzen 11. Insert wedge here Transport und bei abgen Remove transit cover-Abdeckblech Vor dem start entfernen plate & discard before
- Herkules A
- 13. 黑黑
- 14. 477 000

- 1. The nose cone is always 6 Schrauben zu befestigen to be fastened by means of the 6 screws
- 2. Nicht auftreten 2. Not to be stepped on

- Storage cradle A Double cradle
- 4. Black square-no writing
 - Long trolley N
- on underside before use
- Doppelpallung Double cradle
- (Pallungsabstand b2) (cradle spacing b2)

 - 9. Chock
 - 10. Do not touch
 - launching
- 12. Abstellpallung A 12. Storage cradle A
 - 13. Data plate: 2 white bars w/ 3 lines of writing
 - 14. serial no.; either/or both locations

