MAC 188E PROTOTYPE

TOUR OF THE UNITED STATES FROM 12 JUNE THROUGH 8 JULY 1964

MCDONNELL



MAC 188E PROTOTYPE STOL TRANSPORT



MCDONNELL

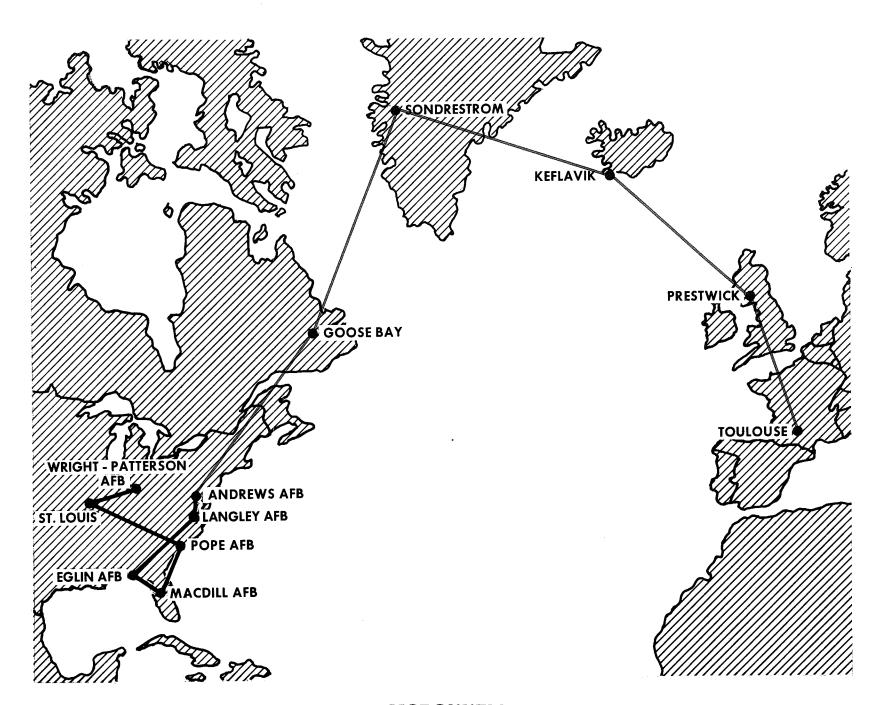
INTRODUCTION

This report presents a brief summary of the tour of the MAC 188E Prototype (Breguet 941.01) to selected USAF installations and the history of the development of this outstanding STOL aircraft. This tour was sponsored by the USAF, The French Air Ministry, McDonnell and Breguet. The principal purpose of the tour was the conduct of a limited operational suitability evaluation of this aircraft by the Air Force's Special Air Warfare Center at Eglin AFB.

The MAC 188E has evolved from three years of close co-operation between Breguet and McDonnell on the 941 aircraft. The 188E will be wholly American produced and qualified to U. S. Military specifications.

IT SHOULD BE NOTED THAT THE AIRCRAFT USED FOR THIS TOUR IS A PROTOTYPE AND THE ONE AND ONLY AIRCRAFT FLYING OF ITS TYPE.

TOUR ROUTE



ITENERARY

June	6	Ferry—Toulouse, France to Prestwick, Scotland
	7	Ferry-Prestwick to Keflavik, Iceland to Sondrestrom, Greenland
	8	Ferry—Sondrestrom to Goose Bay, Labrador
	9	Ferry-Goose Bay to Dulles International Airport
	10-11	. Routine maintanence and aircraft preparation
	12	Familiarization Flights, Andrews AFB
	13	Familiarization Flights, Langley AFB
	15	Demonstration and familiarization flights to Special Air Warfare Center, Hurlburt Field
	17-27	Limited perational Suitability Evaluation by 1st Combat Applications Group, SAWC, Eglin AFB
	29	Demonstration and familiarization flights, MacDill AFB
July	1	Demonstration and familiarzation flights, Pope AFB Demonstration, Simmons AAF, Fort Bragg
	2	Familiarization flights, Pope AFB
	4	Demonstration, St. Louis
	7	Familiarization flights, St. Louis
	8	Demonstration and familiarization flights, WPAFB

DEMONSTRATION FLIGHT PLAN

TYPICAL CONDITIONS

Take-off Gross Weight, lbs.	41,000
Payload (M-37 weapons carrier and personnel) lbs	. 7000
Temperature, °F	90

FLIGHT PLAN

Take-off from sod field in less than 700 ft.

Cruise speed fly-by at 210 kts. and 300 ft.

STOL configuration (85° flaps, gear down) pass at 400 ft. demonstrating high rate of roll at 50 kts.

STOL landing on sod with less than 500 ft. ground roll

Lower cargo ramp and discharge payload in less than one minute

STOL take-off followed immediately by 300 ft. radius turn at 50 kts.

Simulated cargo extraction at 60 kts. at less than 10 ft. altitude

Conversion from landing approach with 800 ft./min. sink to wave-off at 600 ft./min. rate of climb in less than 2 sec.

STOL landing - Touchdown point repeatable within ±25 ft.

Taxi turns with outer wing tip turning on 50 ft. radius

Demonstration of vertical self jacking feature to facilitate cargo unloading (18 in. vertical displacement and 5° tail down floor tilt.)

Two-engine-out characteristics were demonstrated in flight with no change in flight characteristics. With engine-out, all four propellers operate due to cross-shafting.



FLIGHT ACTIVITY SUMMARY 6 JUNE THRU 8 JULY 1964

FLIGHTS	
DEMONSTRATION	5
FAMILIARIZATION	39
EVALUATION (SAWC)	40
FERRY	17
TOTAL FLIGHTS	101
FLIGHT HOURS	
DEMONSTRATION	55
FAMILIARIZATION16:	30
EVALUATION (SAWC)	25
FERRY)0
TOTAL FLIGHT HOURS	100:50
STOL LANDINGS	
HARD SURFACE 16	58
SOD	⊋ 5
CLAY	9
SAND	31*
PSP	2
TOTAL LANDINGS	305

^{*}Take off and landings were successfully made in sand, sufficiently soft, such that 14 inch ruts were produced.

SAWC LIMITED OPERATIONAL SUITABILITY EVALUATION

17-27 JUNE 1964

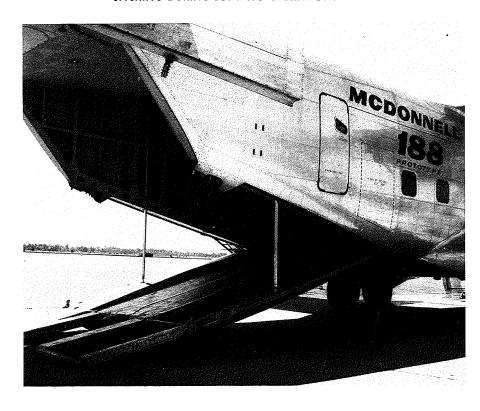
PURPOSE OF EVALUATION

- 1. Determine pilot training requirement for check-out
- 2. Determine performance and handling characteristics from hard surface, sod, sand, clay and pierced steel planking
- 3. Determine instrument flying characteristics
- 4. Investigate unusual design features
- 5. Investigate cargo loading and unloading features
- 6. Investigate acceptability of emergency procedures

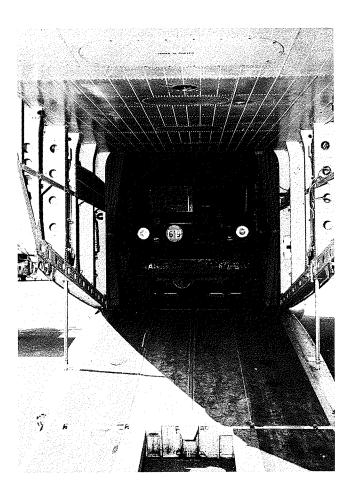
RESULTS (35:25 flight hours, 40 flights, 200 landings)

- 1. Time required for complete pilot check-out is approximately 4 hours
- 2. Varying field surface conditions have essentially no effect on performance or handling characteristics during take-offs and landings. (For example the maximum difference in take-off distance for the various field surfaces was only 30 ft.)
- 3. Instrument flight including short field landing and take-off is improved compared to conventional aircraft
- 4. No special AGE was required for cargo-loading operations. Cargo loading features and turn around time were considered outstanding.
- 5. Separate flights involving the feathering of an inboard and an outboard propeller showed aircraft control and performance to be excellent. One two-engine take-off was made without effect on aircraft handling characteristics.
- 6. Night STOL operations with and without lights are easily accomplished
- 7. Slow speed maneuvering flight at low altitude is accomplished with safety and provides a high degree of pilot confidence

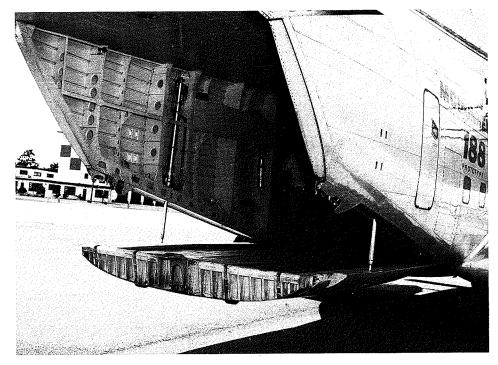
HYDRAULIC ACTUATORS ON RAMP PROVIDE AFT FUSELAGE JACKING DURING LOADING OPERATIONS



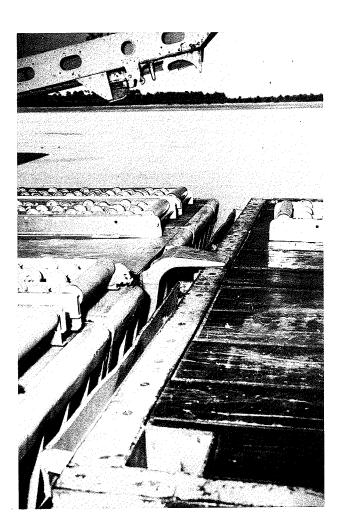




HYDRAULIC ACTUATORS LOCK RAMP IN THE HORIZONTAL OR ANY DESIRED POSITION

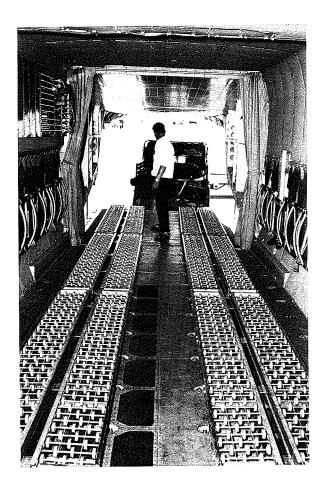


HOOK ON AFT END OF TAIL RAMP RESTS ON TRUCK
BED AND PREVENTS RELATIVE MOTION OF RAMP
AND BED DURING LOADING OPERATIONS



2500 POUND BOX BEING ON-LOADED

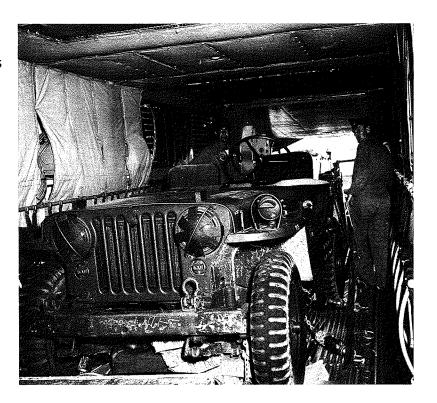




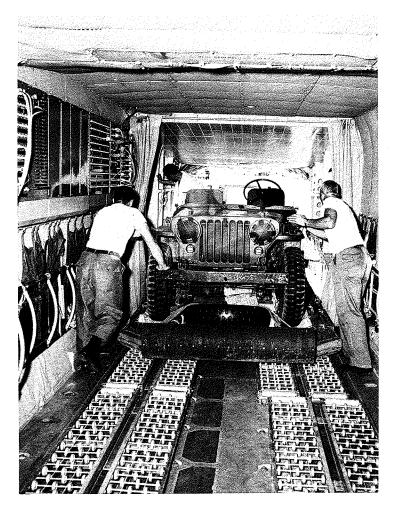
"ROLLER SKATES" INSTALLED. TROOP SEATS STORED

PALLETIZED LOAD PROVIDES PERSONNEL ACCESS





"JEEP" AND 3/4 TON TRUCK LOADED



PALLETIZED "JEEP" BEING OFF-LOADED

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French Air Attache

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Lt. Duc

1st Air Commando Wing

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Senior RAF officer, ASD

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L/Col. D. Brett	Chief Standardization and Evaluation

MCDONNELL

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SESOM, ASD

Chief Special Warfare Div., TAC

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AFSC Commander 775 TCS

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SAWC
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L/Col. F. Oettinger

Brig. Cmdr. 2nd Brig 82nd Airborne Div.

Doctrine & Requirements, Director of

Army CDC Liaison Officer, STRICOM

Div. Inspector General, 82nd Airborne

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Maj. M. E. Jamison

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CDC Liaison Officer, AVCOM

HQ & HQ 82nd Airborne

82nd Airborne

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UNITED STATES NAVY

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Cmdr. R. D. Nye

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B/Gen. L. E. English

Deputy Director of Plans J5, STRICOM

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Engineering Manager, MAC

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Ops Analysis, TAC

Vice President, General Engineering, MAC

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Vice President, Advanced Product Planning, MAC RD & E, AVCOM Proj. Aero Engr. - V/STOL, MAC ASZXC, ASD Field Service Representative, MAC CV7 Project Office, AVCOM Director Development, AVCOM

RD & E, AVCOM President, MAC

Mgr. Advanced Engr. V/STOL, MAC CV7 Project Office, AVCOM Director New Programs, ASD Chairman, and Chief Executive Officer, MAC Vice President, General Manager, Combat Aircraft, MAC Chief Aerodynamics Engineer, MAC

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${\tt Mr.}$	R_{ullet}	В.	Sh	ort
${\tt Mr.}$	Cha	rle	es	Slatt
$\mathtt{Mr}.$	G_{\bullet}	\mathtt{B}_{ullet}	Sl	oan
Mr.	\mathbf{F}_{ullet}	I.	St	eele
				phensor
${\tt Mr}$.	C.	R_{ullet}	Th	ielin
Mr.	Gen	e 7	<i>T</i> ar	ble
Mr .	How	arc	l W	ilson

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Project Engineer, MAC
AVCOM
Advance Product Planning, MAC
Project Engineer, MAC
RD & E, AVCOM
SEAEP, ASD
Ops Analysis, TAC
Director Engineering, AVCOM

MAINTENANCE

UTILIZATION	
FLIGHT HOURS PER MONTH	97.5
MAXIMUM FLOWN PER DAY BY SAWC	6.5 HRS.
MANPOWER	
ALL MAINTENANCE WAS PERFORMED BY FOUR BREGUET	_
AND THREE MCDONNELL MECHANICS	7

DEVELOPMENT HISTORY

STOL studies initiated by Breguet	954
First flight of research aircraft, Breguet 940	1958
First flight of operational aircraft, Breguet 941.01June 1	1961
Technical agreement, Breguet/McDonnellFall 1	1961
License agreement, Breguet/McDonnellSpring 1	962
First MAC 188 design	962
MAC Operations Analysis and Cost Effectiveness Studies	963
MAC 188E preliminary design completed ·····Spring 1	963
Breguet production contract from French Air Ministry announcedJune 1	963
U. S. Flight evaluations of 941.01	
NASA/AFFTC (Stability and Control)	963
AFFTC/ASD/NASA (Handling qualities and instrument flying capabilities)	963
AFFTC/ASD/U. S. Army (Off-runway operations)	963
European demonstration tour October 1	963
Depart Toulouse, France for U. S. tour	964

COMPARISON OF AIRCRAFT CHARACTERISTICS

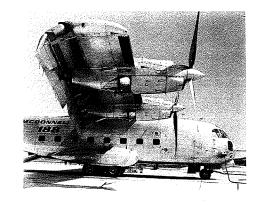
	BREGUET 940.01	MCDONNELL 188E
SHAFT HORSEPOWER (EACH)	1165	1465
MAXIMUM GROSS WEIGHT	52,920 LBS.	58,422 LBS.
USEFUL LOAD (PAYLOAD + FUEL)	25,000 LBS.	29,357 LBS.
MAXIMUM PAYLOAD	14,300 LBS.	17,500 LBS.
CRUISE SPEED @ 10,000 FT	180-215 KT\$.	195-255 KTS.
APPROACH SPEED	55 KTS.*	55 KTS.*
GROSS WEIGHT TO CLEAR 50 FT. IN 1000 FT.		
STD DAY, SEA LEVEL	41,400 LBS.	46,100 LBS.
103°F, SEA LEVEL	35,600 LBS.	42,000 LBS.
MAXIMUM LANDING WEIGHT	44,000 LBS.	51,600 LBS.
LANDING DISTANCE OVER 50 FT. AT		
MAXIMUM LANDING WEIGHT	730 FT.	820 FT.
FERRY RANGE WITH AUXILIARY FUEL	1800 N.M.	3040 N.M.
WINGSPAN	76.1 FT.	76.7 FT.
LENGTH	72.9 FT.	77.9 FT.
HEIGHT	30.7 FT.	30.9 FT.
CABIN		
WIDTH	8.0 FT.	8.5 FT.
HEIGHT	7.4 FT.	7.4 FT.
LENGTH	34.0 FT.	36.6 FT.

The 188E is designed to carry over 90% of all the wheeled and tracked vehicles of a ROAD Airborne Division.

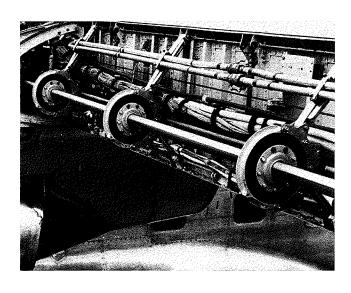
^{*} Approach speed is essentially independent of gross weight because of the "POWERED LIFT" aspect of this aircraft's design.

MAC 188 SALIENT DESIGN FEATURES

- DEFLECTED SLIPSTREAM—WING WHOLLY IMMERSED IN PROPELLER SLIPSTREAM.
- CROSS-SHAFT INTERCONNECTION OF THE FOUR PROPULSION UNITS.
- HIGHLY DEFLECTED, FULL-SPAN, TRIPLE-SLOTTED FLAPS.
- RELATIVELY LARGE DIAMETER PROPELLERS DESIGNED FOR HIGH STATIC THRUST.
- STICK ACTUATED DIFFERENTIAL PITCH BETWEEN OUTBOARD PROPELLERS FOR LOW SPEED CONTROL.
- ONE PROPELLER CONTROL LEVER FOR ALL FOUR PROPELLERS.
- ONE THROTTLE TO CONTROL ALL FOUR ENGINES.
- "TRAILING ARM" LANDING GEAR FOR ROUGH FIELD OPERATION.



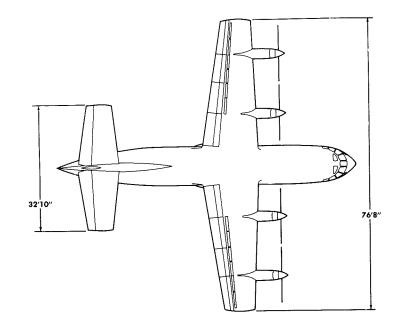
Inboard flaps deflected 98°, outboard 72°

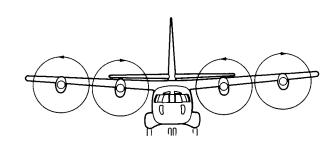


Propulsion system cross-shafting

TRANSPORT DIMENSIONAL COMPARISON

MODEL	WING SPAN	LENGTH	HEIGHT
188E	76.7′	77.9′	30.9
C-130B	132.6′	97.8′	38.0′
C-123	110.0′	76.2′	34.5′
CV-2A	96.4′	72.6′	31.8′
CV-7A	96.0′	76.9′	28.7′



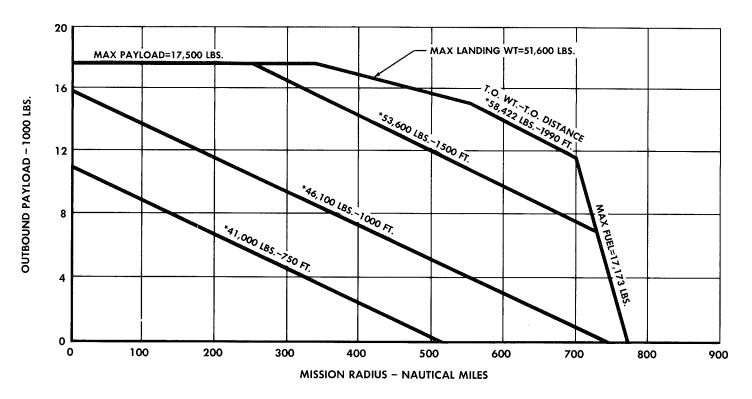




MAC 188E PAYLOAD vs RADIUS 10,000 FT. CRUISE

4 ENGINES CRUISE • ENGINES 1465 HP EA.
RETURN ½ OUTBOUND PAYLOAD • CRUISE SPEED = 200 KNOTS





*INITIAL TAKE-OFF CONDITION NOTE: CRUISE ON 3 ENGINES $_{\textcircled{@}}$ 200 KTS. INCREASES RADIUS APPROX. 6.5%