

COMPETITION SENSITIVE

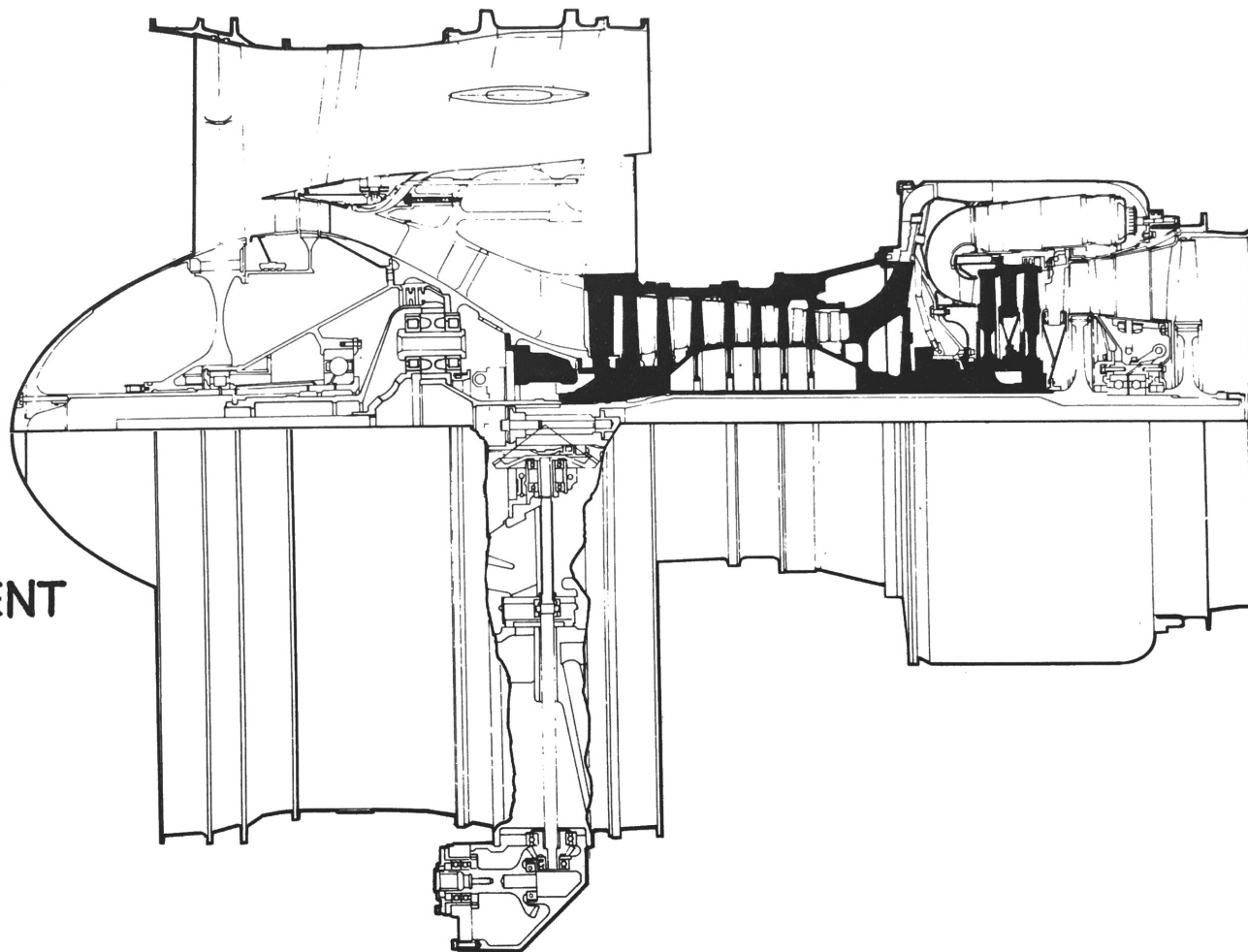
NB 72-325
20 DECEMBER 1972

A-9
SOURCE SELECTION
ADVISORY COUNCIL
BRIEFING
20 DECEMBER 1972

COMPETITION SENSITIVE

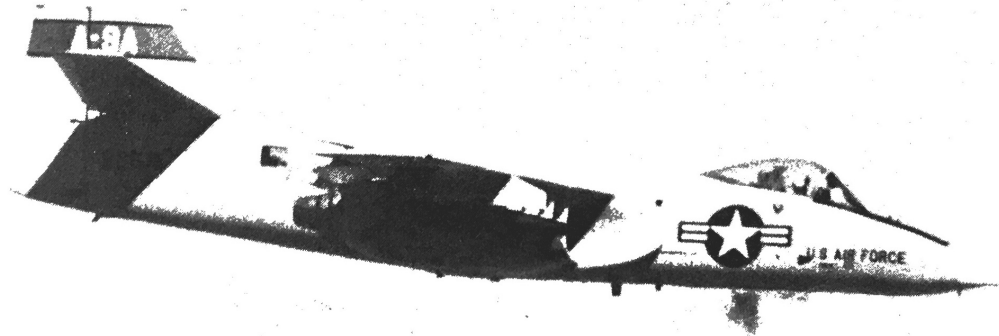
YF102 ENGINE

- DEVELOPED FROM T-55
- GEAR FAN MODULE ADDED
- LOW DEVELOPMENT COST



51939A

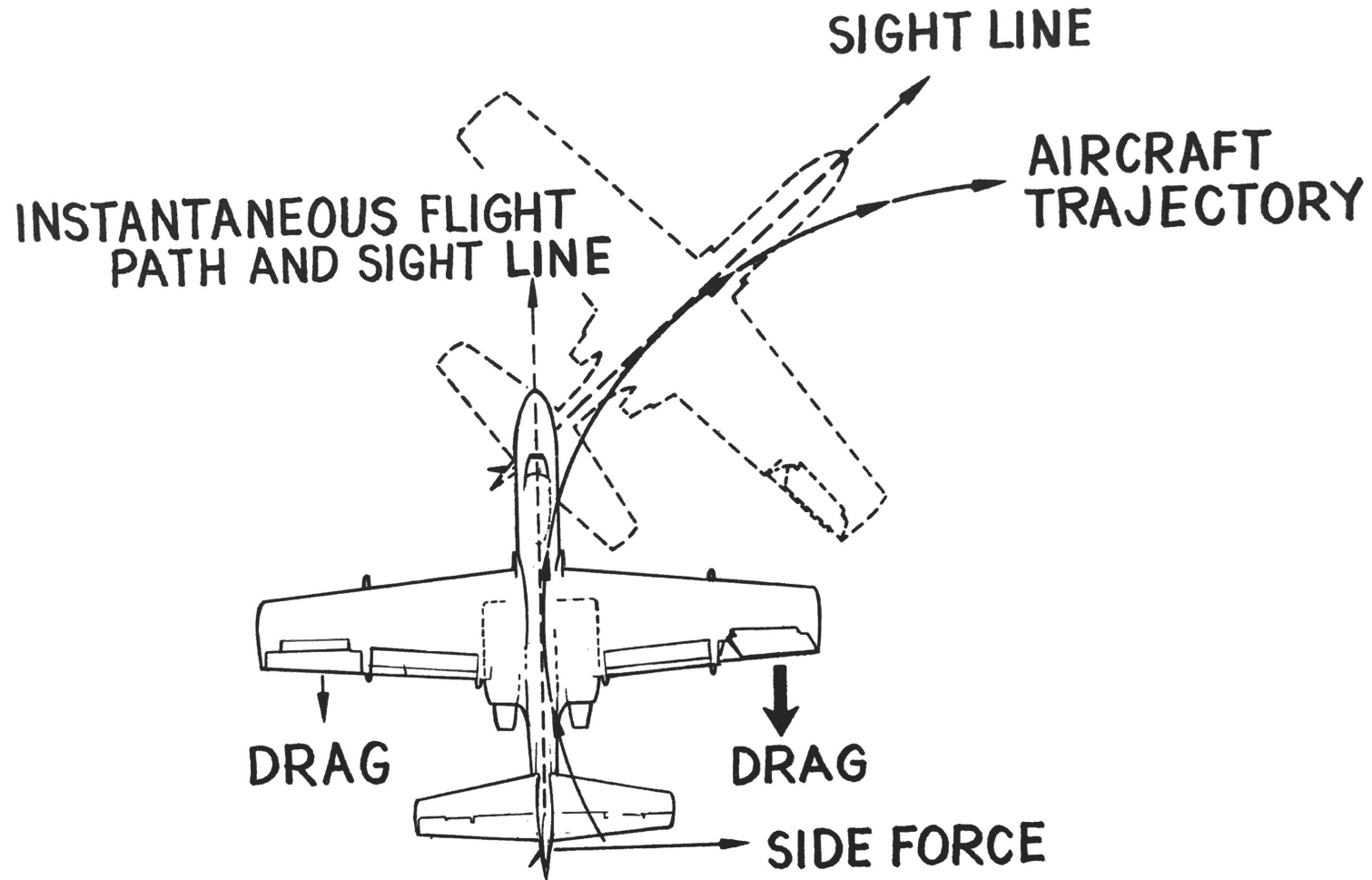
A-9 GUN GAS DEFLECTOR



NO PITCH UPSET
NO INGESTION OF GUN GAS OR DEBRIS
MINIMUM FLASH

DESIGN VERIFIED

SIDE FORCE CONTROL

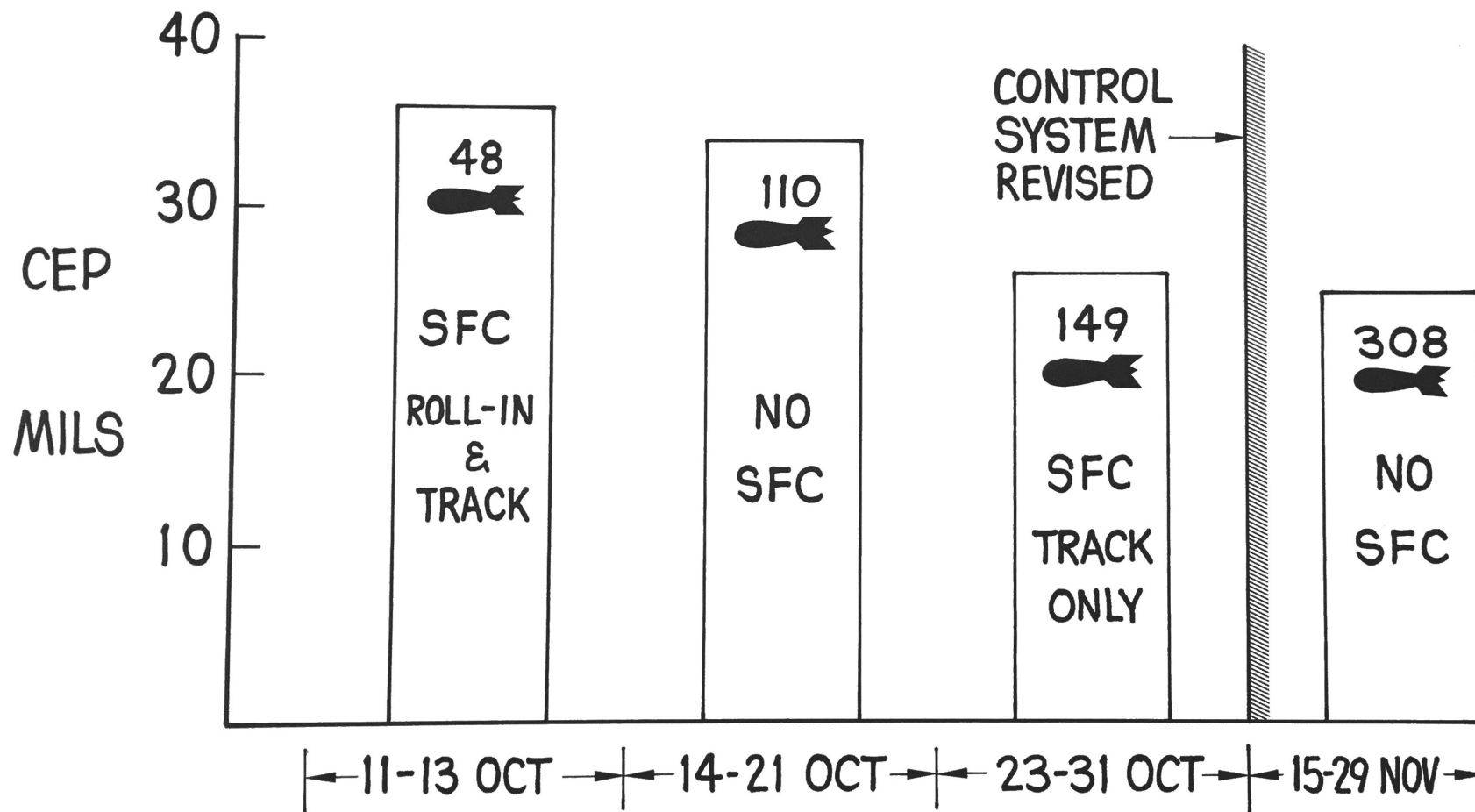


SIDE FORCE CONTROL

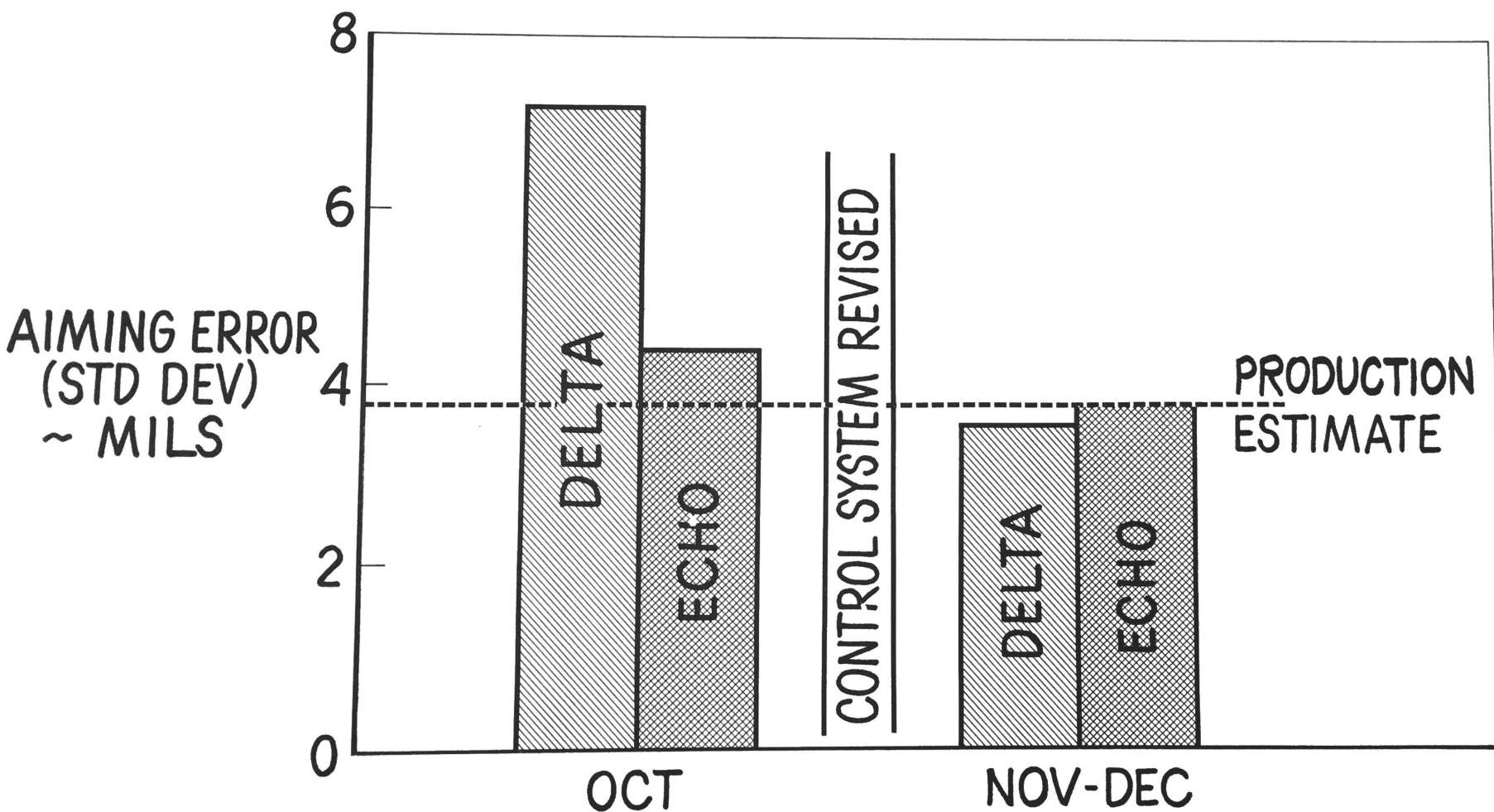
- MECHANIZATION NON-OPTIMUM
 - PITCH & ROLL UPSETS
 - PITCH DAMPER & INTERCONNECTS NOT OPERATIONAL
- ACCELEROMETER YAW SAS PREFERRED
 - GRAVITY TERM ($q_f \cos \theta \sin \phi$) NOT INCORPORATED
 - ADVERSE YAW IN TURNS
 - WASHOUT CIRCUIT DEGRADED DAMPING
- POTENTIAL PAYOFF STILL VALID

A-9 BOMBING ACCURACY

ALL PROFILES



A-9 STRAFING ACCURACY



A-9 ACCURACY DEMONSTRATED

BOMBING > MET PROTOTYPE ESTIMATE

FLYOFF	20-22 MILS
WILDCARD	13-14 MILS

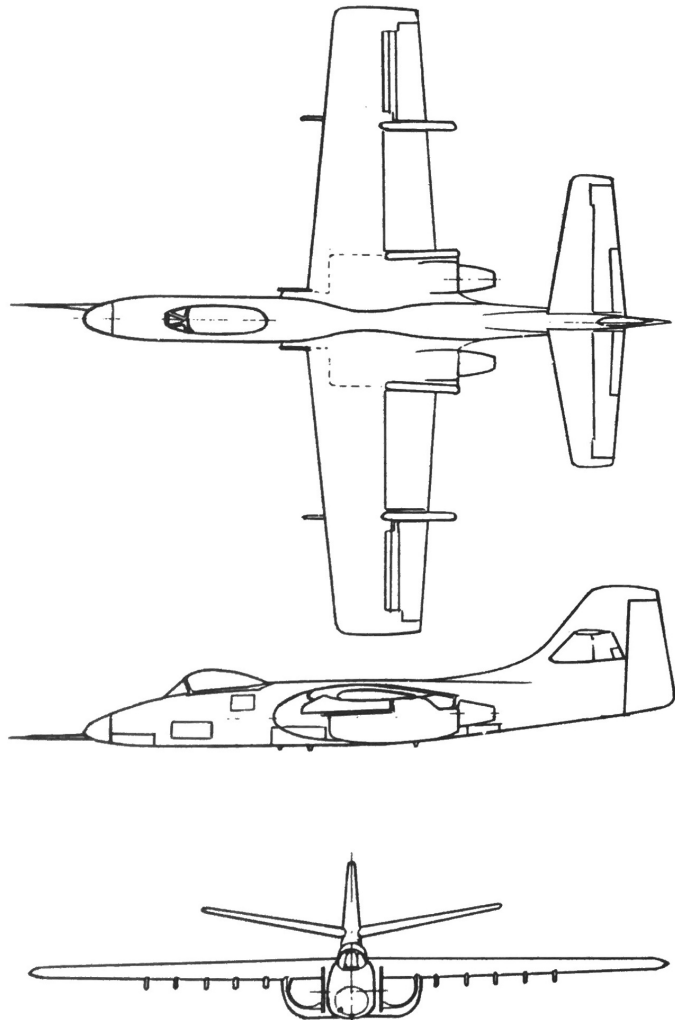
STRAFING > MET PRODUCTION ESTIMATE

FLYOFF	3.5 - 4.0 MILS
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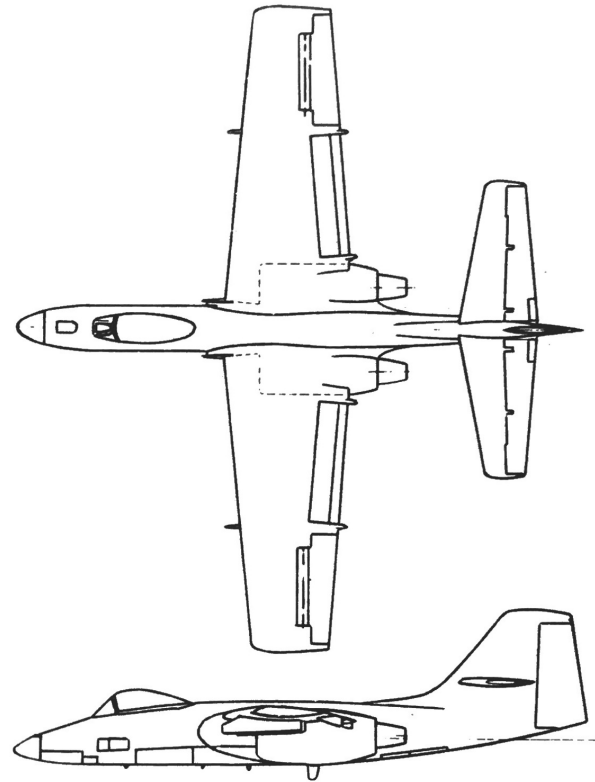
FDL S/V TEST SUMMARY

- SHOTS INTO FUEL 20 — NO FIRES
- SHOTS INTO DRY BAYS 13 — HAZARD ZONE
SMALLER THAN
PREDICTED
- FIRES 4 (EXPECTED 5)
- A_v BETTER THAN USAF ESTIMATES (MAY 1971)

SIMILARITY OF PROTOTYPE & PRODUCTION CONFIGURATIONS



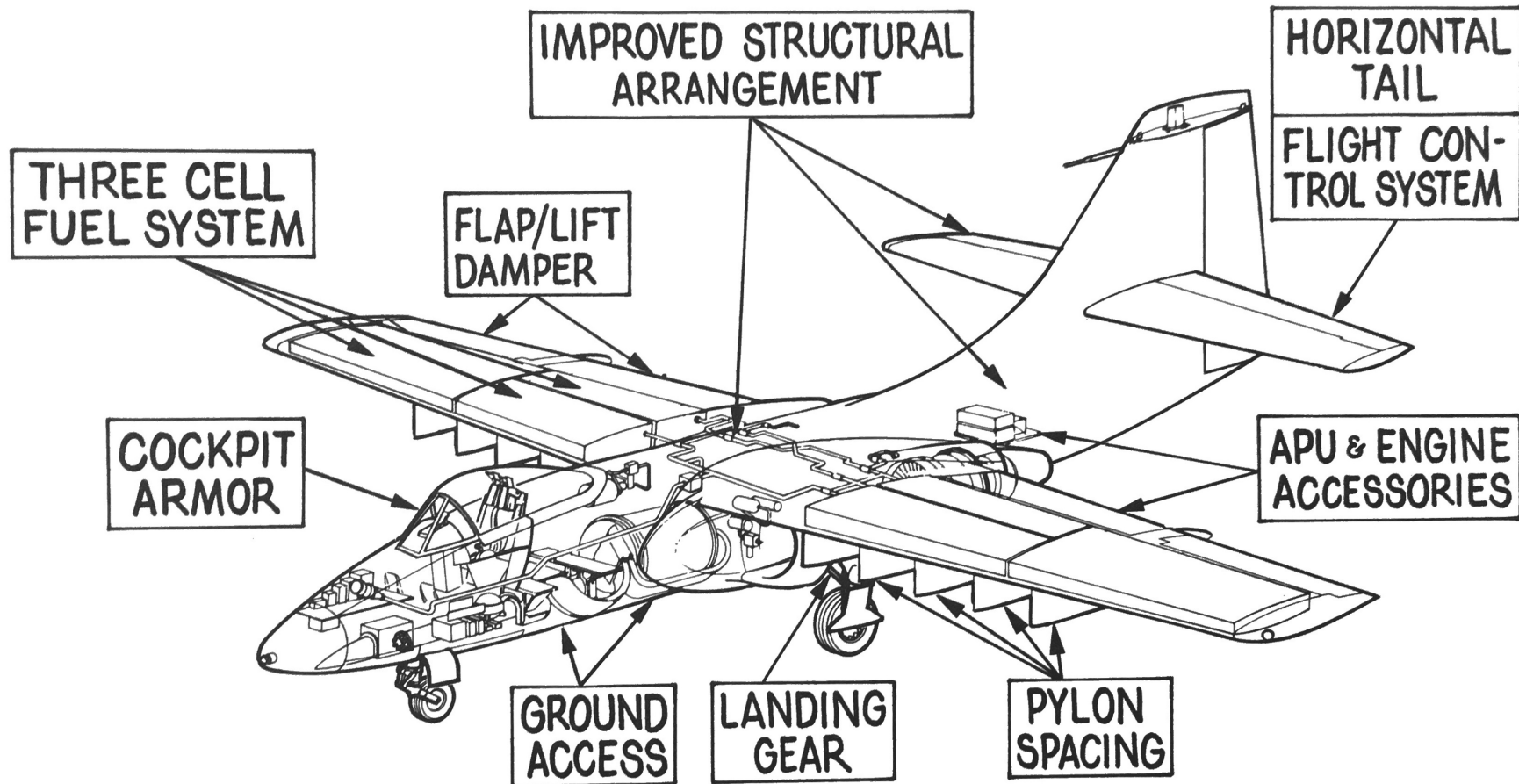
PROTOTYPE



PRODUCTION

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A-9 PRODUCTION IMPROVEMENTS RESULTING FROM CPP

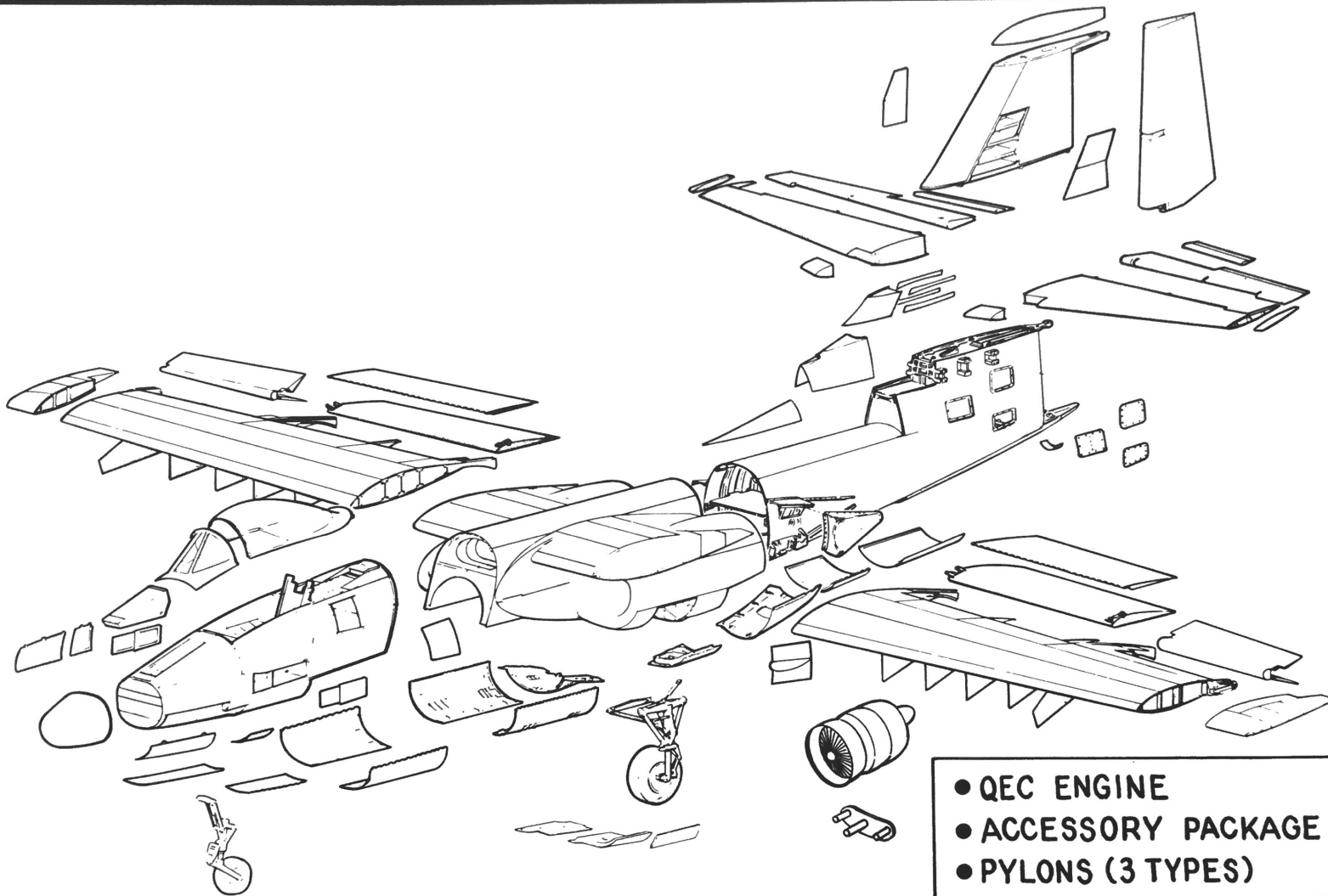


COST - PERFORMANCE - EFFECTIVENESS

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A-9 PRODUCTION BREAKDOWN



51960A

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A-9 COMBAT PERFORMANCE

(TROPICAL DAY)

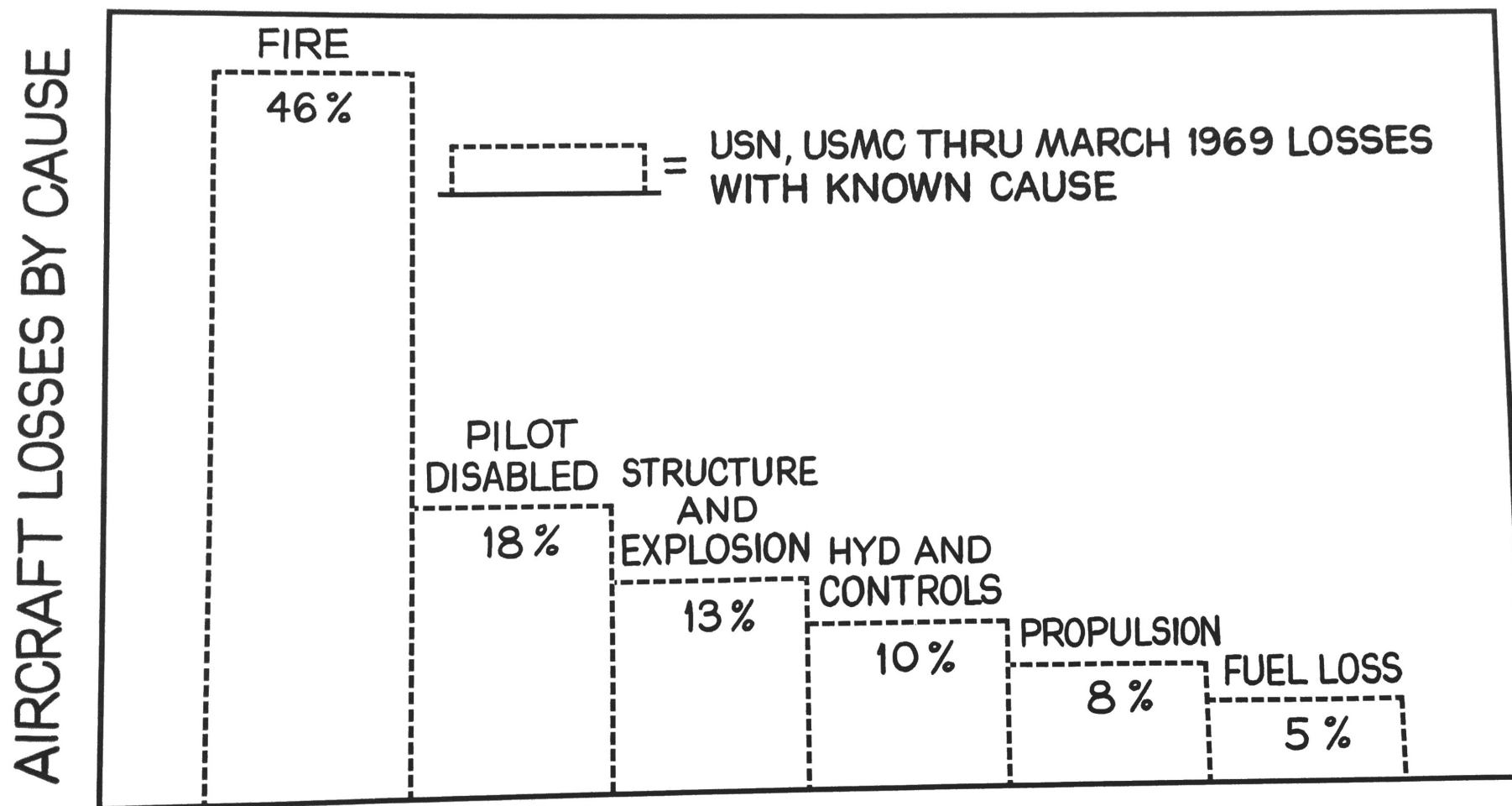
	A-X GOAL	A-9 PROTOTYPE	A-9 PROD
MANEUVERABILITY(5000 FT)			
– INSTANT. G 300 KTAS	5.0	6.3	6.3
– SUSTAINED G 275 KTAS	3.5	3.7	4.0
RATE OF CLIMB, FT/MIN	NOT SPECIFIED	4,500	5,150
MAX SPEED			
– CLEAN (SL) KTAS	400	390	410

A-9 AIRFIELD PERFORMANCE

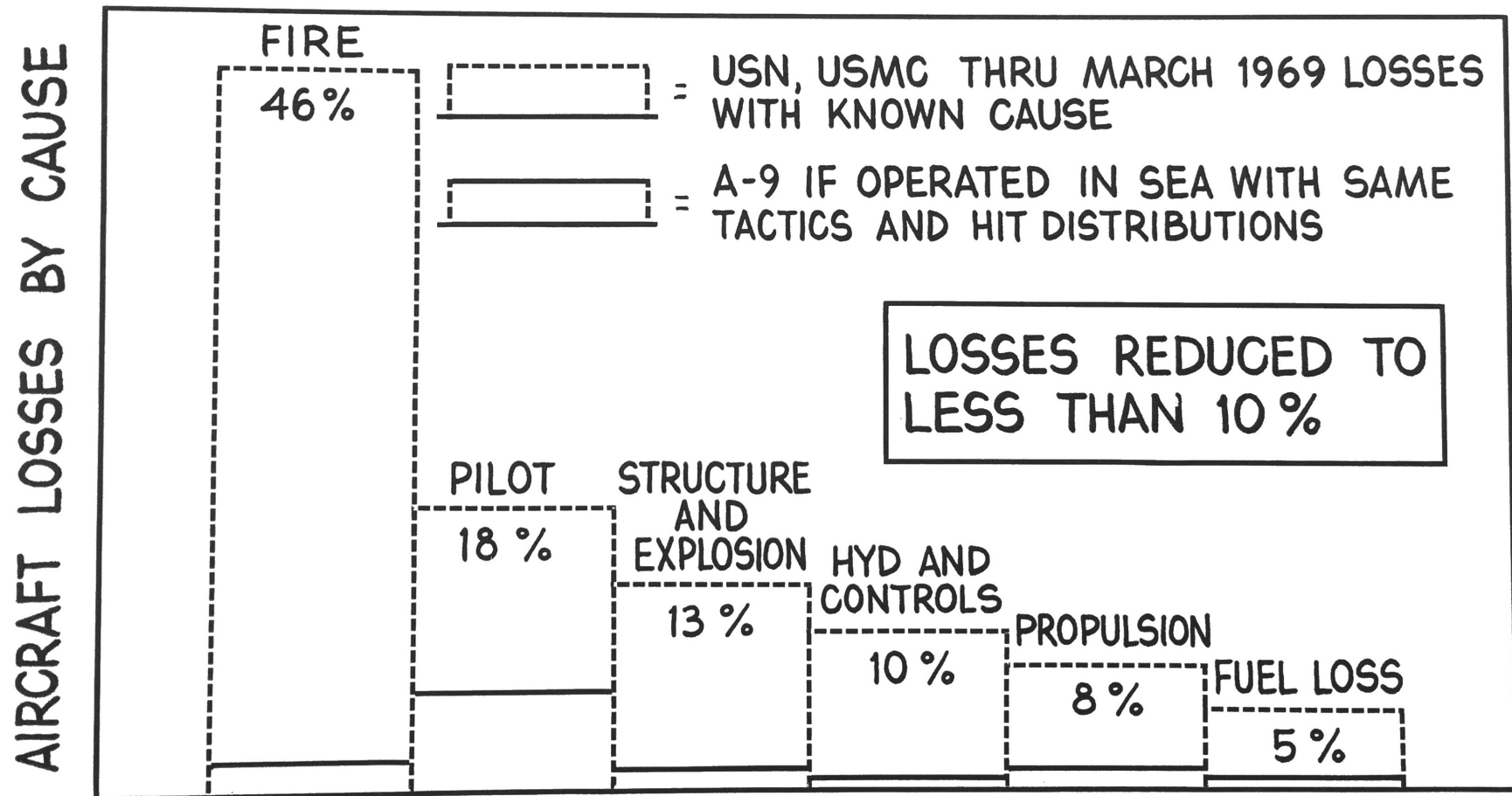
(TROPICAL DAY)

		A-X GOAL	A-9 PROTO	A-9 PROD
FORWARD AIRSTRIP (CPP RULES) SL				
— TAKEOFF	FT	1000	750	780
— LANDING	FT	1000	950	1000
MAX TAKEOFF-WEIGHT SL				
— TAKEOFF	FT	4000	3900	2900
— LANDING	FT	4000	2100	1800
— SINGLE ENGINE CLIMB GRADIENT	%	1.0	0.6	1.0

AIRCRAFT LOSSES BY CAUSE



EFFECT OF A-9 PROTECTION MEASURES



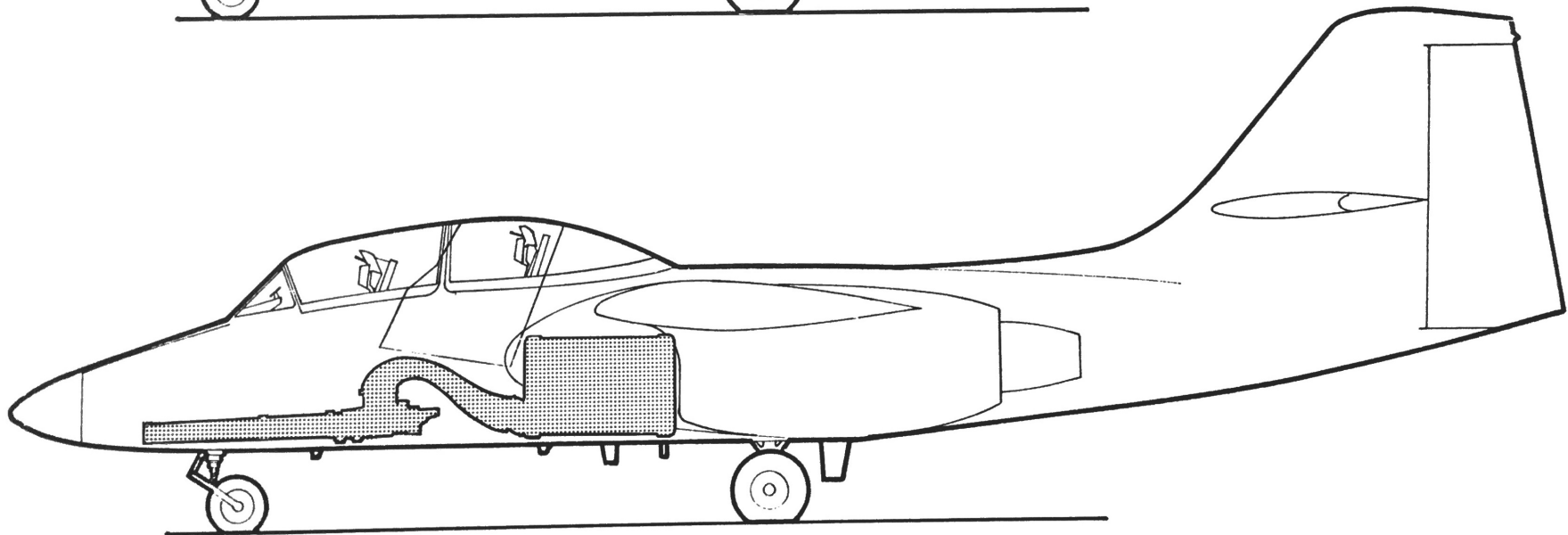
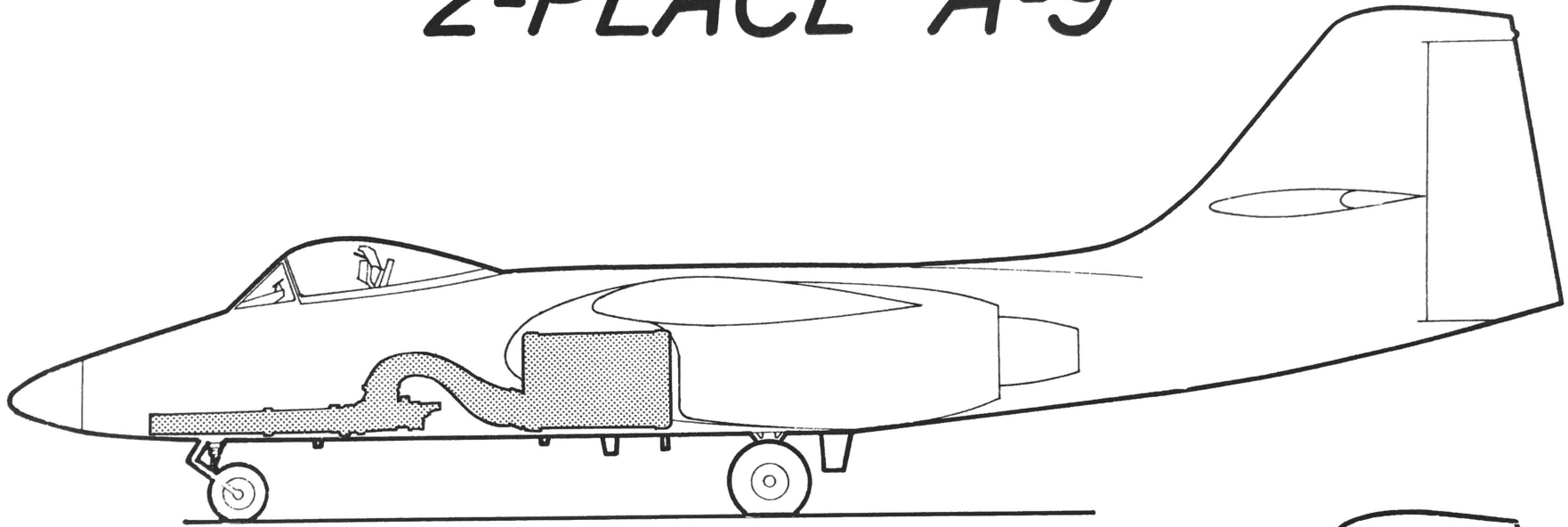
COMBAT PERFORMANCE

(TROPICAL DAY)

		A-9 PROD	F-102 GROWTH
BASIC DESIGN WT	LB	25,950	26,070
MANEUVERABILITY (5000 FT)			
– INSTANT. G	300 KTAS	6.3	6.3
– SUSTAINED G	275 KTAS	4.0	4.2
RATE OF CLIMB, FT/MIN		5,150	6,000
MAX SPEED			
– CLEAN (SL)	KTAS	410	420
– 6 MK-82's, 5000 FT	KTAS	400	410

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2-PLACE A-9



52079A

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TEN-YEAR O&S COST COMPARISON

COST ELEMENT	COST ~ MILLIONS OF 1970 \$		
	A-9	A-7 ⁽¹⁾	F-4 ⁽²⁾
SPARE ENGINES & LRU's	106.9	190.3	144.2
ON & OFF EQUIP. MAINTENANCE	574.4	1059.0	1303.9
POL	84.8	176.9	398.7
PECULIAR SUPPORT	47.1	540.0	248.9
NEW ITEM MANAGEMENT	7.8	14.4	25.8
TOTAL O&S COST	821.0	1980.6	2121.5

(1) ESTIMATE BASED ON AFM 172-3

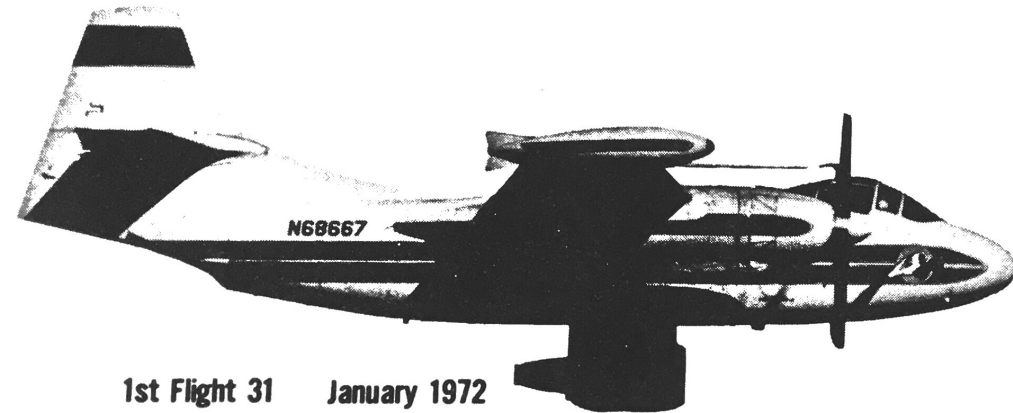
(2) REF: 1. USAF F-4 O&S COST

2. AFM 172-3 USAF COST AND PLANNING FACTORS

3. AFLC DO24F102-N3 AIRCRAFT ENGINE EXPOSURE

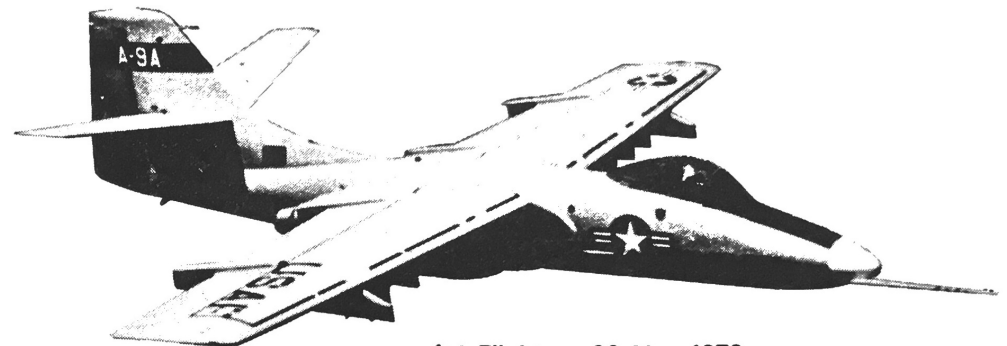
YF102 ENGINE FLIGHT EXPERIENCE

AJ-2 FLIGHT TEST A/C



1st Flight 31 January 1972

NORTHROP A-9

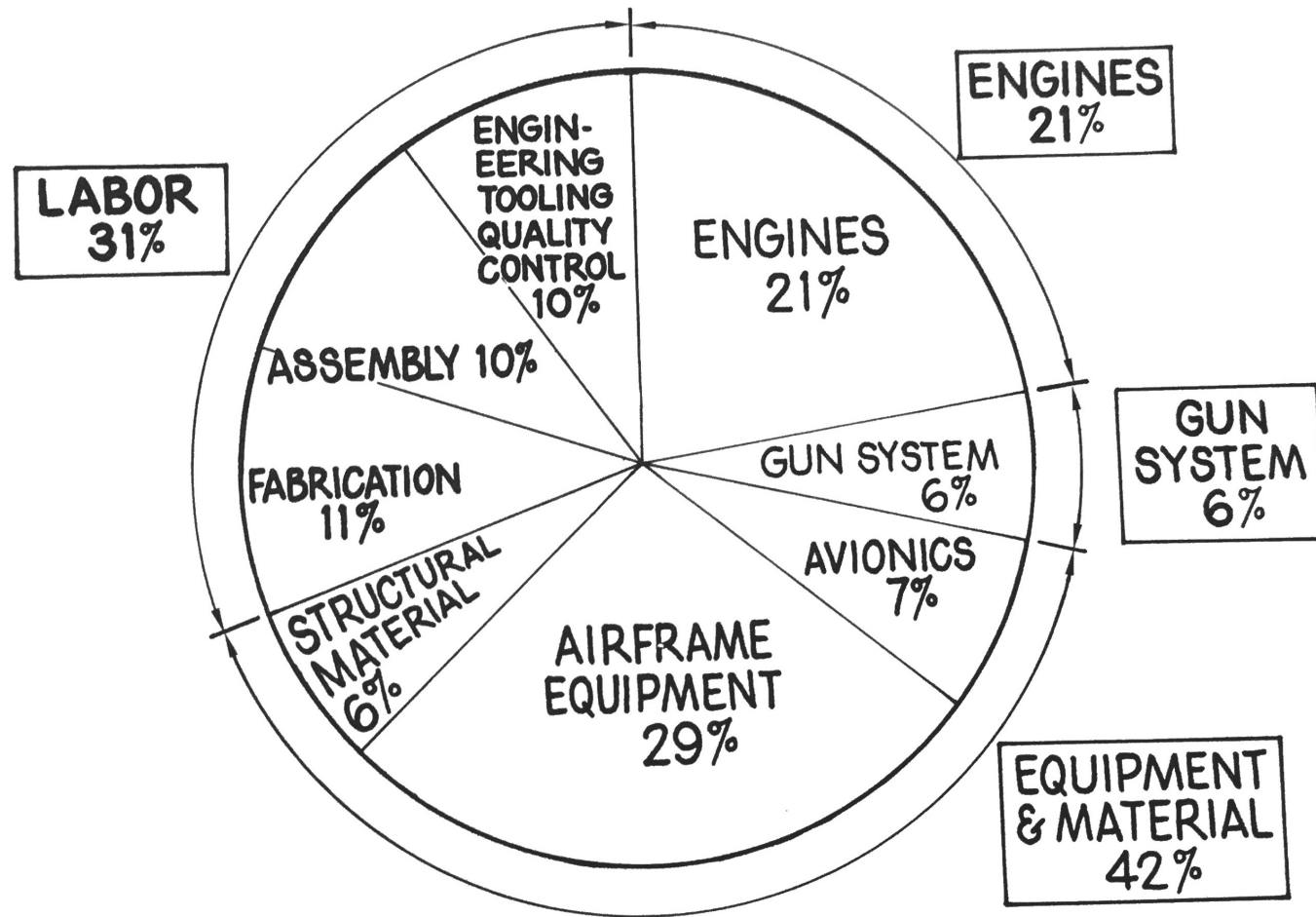


1st Flight 30 May 1972

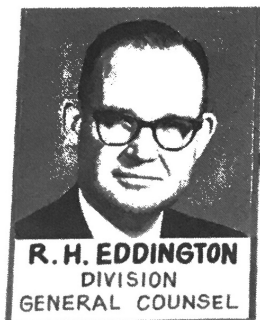
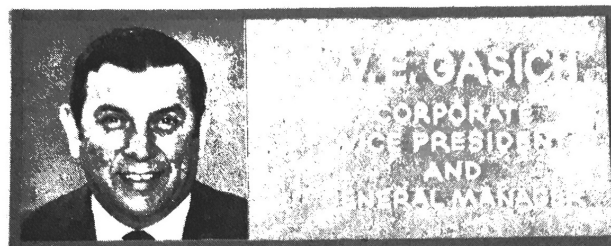
ENGINE NO.	TIME
YF01	199
YF02	180
YF03	136
YF04	150
YF05	143
YF06	121
Total	929 Hours

PROBLEM AREAS	SOLUTIONS
POWER TURBINE BLADE CRACKS (2)	NEW MATERIAL/CONFIGURATION ON TEST FOR QT
GEAR SUPPORT RING FAILURE (1)	HIGH STRESS CONDITION ELIMINATED - ALL ENGINES RETROFITTED
LUBE SYSTEM	IMPROVED SEPARATOR AND TRANSFER TUBE RETROFITTED. QT HOSE CLAMP IMPROVED.

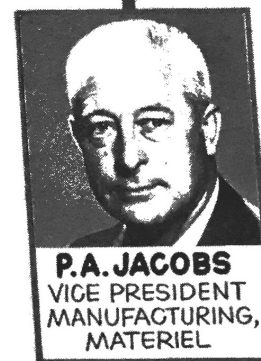
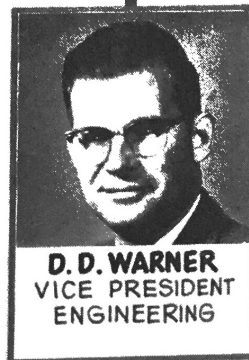
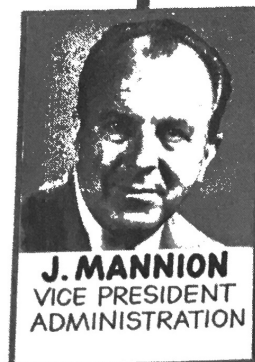
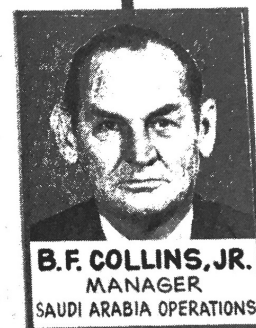
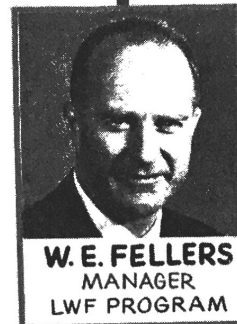
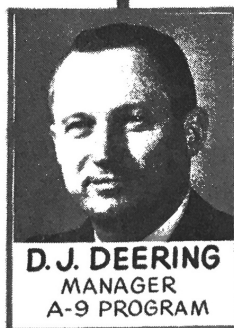
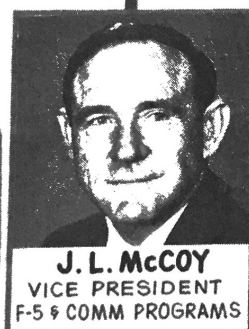
A-9 FLYAWAY COST ELEMENTS



AIRCRAFT DIVISION MANAGEMENT



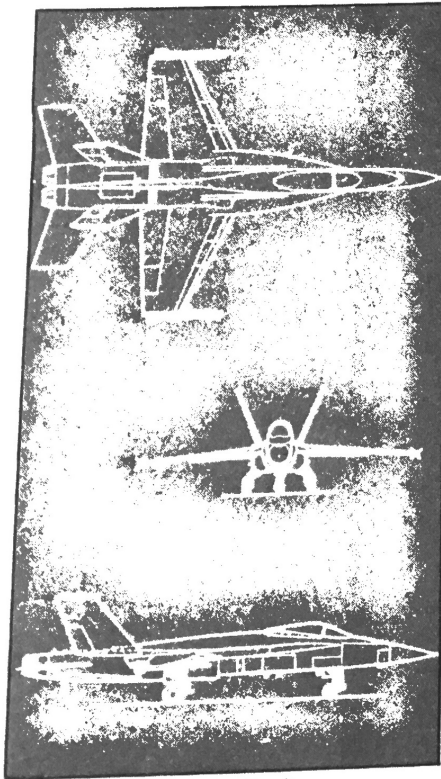
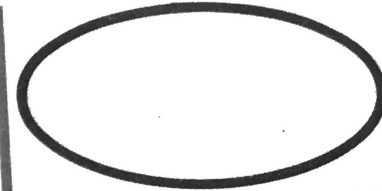
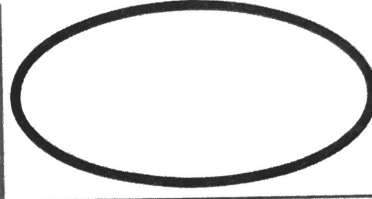
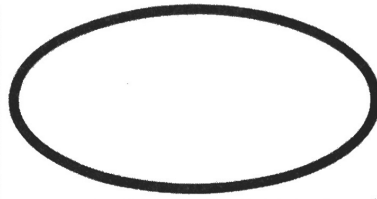
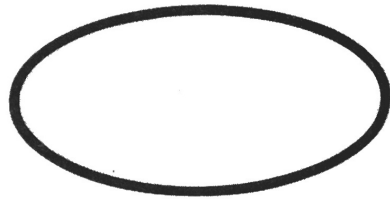
PROGRAM
ORGANIZATION →



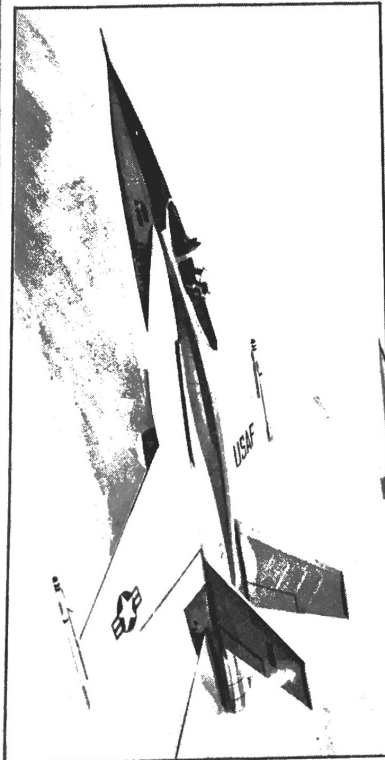
X-127C-1B-1

FUNCTIONAL ORGANIZATION

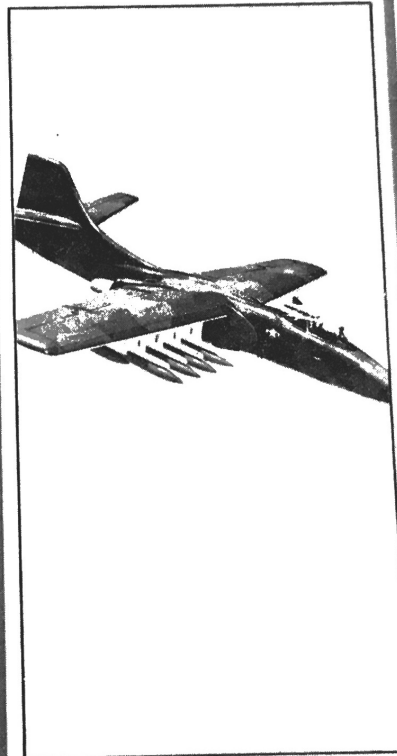
PROJECT PHASES



COBRA



LWF



A-9A



F-5