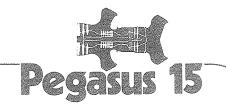


MCDONNELL AIRCRAFT COMPANY

Box 516, Saint Louis, Missouri 63166 - Tel. (314)232-0232







ADVANCED HARRIER CONCEPT

- MEET THE USMC REQUIREMENT FOR SUBSTANTIAL IMPROVEMENT IN V/STOL LIGHT ATTACK CAPABILITY.
- THROUGH EVOLUTIONARY DEVELOPMENT BASED ON THE AV-8A, PROVIDE AN ADVANCED V/STOL AIRCRAFT TO MEET THE NEEDS OF THE USMC, AND USN.
- DEFINE AIRCRAFT CONFIGURATION AND PERFORMANCE AND PROVIDE TEST VERIFICATION OF MINIMUM TECHNICAL RISK.
- ESTABLISH A DEVELOPMENT AND PRODUCTION SCHEDULE, AND ESTIMATE COSTS WHICH CAN BE MET.





. . . "EVOLUTIONARY DEVELOPMENT BASED ON THE AV-8A". . .

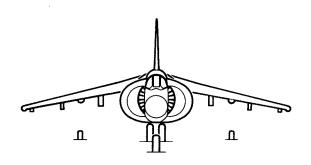
- UPRATE THE PEGASUS ENGINE
 INCREASED THRUST
 IMPROVED TECHNOLOGY
- ADVANCED TECHNOLOGY WING INCREASED RANGE ADDED INTERNAL FUEL IMPROVED MANEUVERING GREATER EXTERNAL LOAD
- AIRCRAFT AND SYSTEMS
 ACCOMMODATE WING AND PEGASUS 15
 INCORPORATE ADVANCED TECHNOLOGY
 PROVIDE FOR ADVANCED MISSION AVIONICS
- ... "AND PROVIDE TEST VERIFICATION
 OF MINIMUM TECHNICAL RISK" . . .

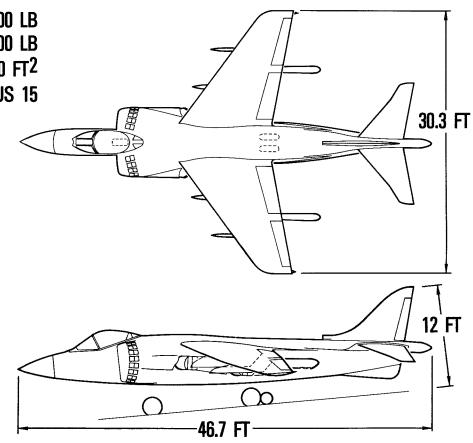




USMC AV-16A

0.W.E. ______13,200 LB
DESIGN TAKEOFF GROSS WEIGHT.... 28,000 LB
WING AREA 230 FT²
ENGINE _____PEGASUS 15

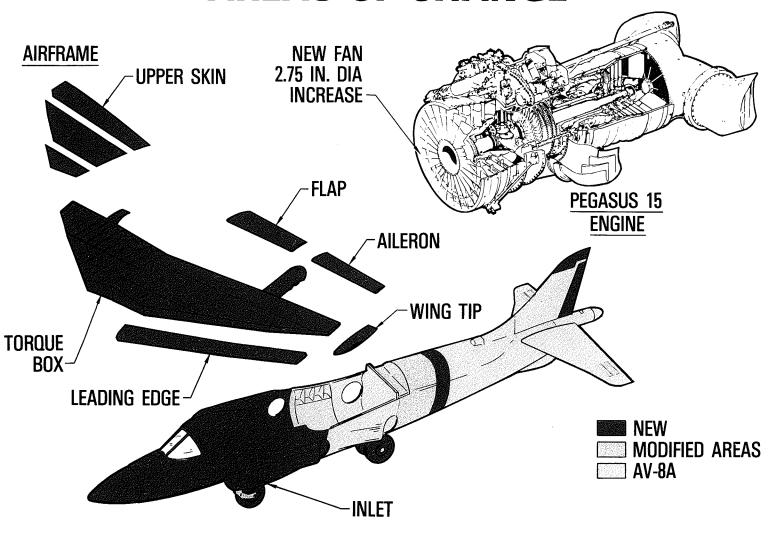








AREAS OF CHANGE

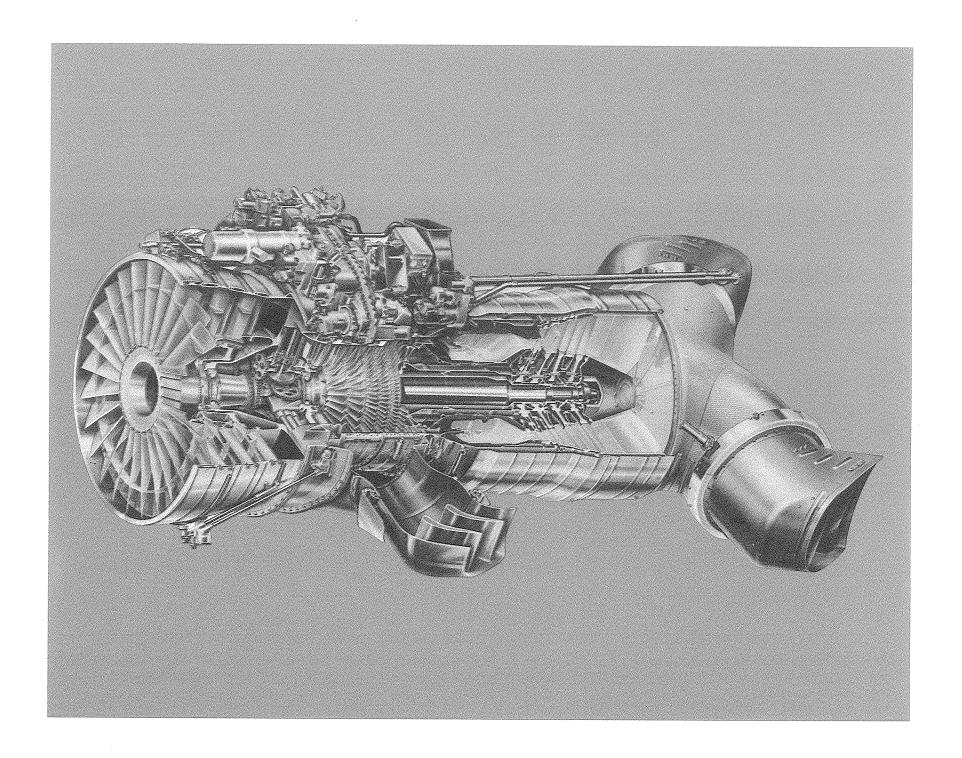






AV-16A ADVANCED TECHNOLOGY FEATURES

- SUPERCRITICAL WING
 - Greater Cruise Efficiency
 - Improved Buffet Boundary for Maneuvering
 - Higher Drag Rise Mach Number
 - Added Internal Fuel Volume
- WEAPON DELIVERY SYSTEM
- LIGHTWEIGHT, RELIABLE ELECTRICAL SYSTEM
 - Power Generating
 - Wiring
- STAINLESS STEEL HYDRAULIC/PNEUMATIC TUBING
- CARBON WHEEL BRAKES
- MATERIALS
 - Titanium
 - Composites







PEGASUS 15 DEVELOPMENT OBJECTIVES

- 15% THRUST INCREASE
- IMPROVE THRUST/WEIGHT RATIO
- LOW COST/LOW RISK COLLABORATIVE PROGRAM





MAJOR COMPONENT IMPROVEMENTS

FAN

- INCREASED AIRFLOW AND EFFICIENCY

 HIGH PRESSURE COMPRESSOR INCREASED EFFICIENCY THROUGH
 AERODYNAMIC IMPROVEMENTS

COMBUSTOR

 IMPROVED HEAT RELEASE AND COOLING SCHEMES

• TURBINE

IMPROVED COOLING FOR HIGHER
 TEMPERATURES

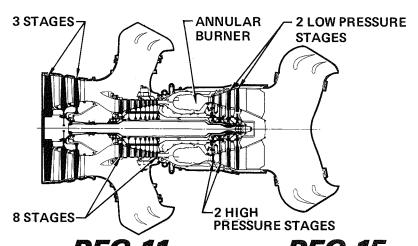
NOZZLES

- FULL ENVELOPE THRUST VECTORING





PEGASUS CHARACTERISTICS



SHORI LIFI WEI RAIINGS	PEG 11	PEG 15
THRUST (LBS)	***************************************	+15%
INLET FLANGE DIAMETER	+ 2	2.75 (INCHES)
TOTAL AIRFLOW (LB/SEC)	444	476
BYPASS RATIO	1.36	1.36
FAN PRESSURE RATIO	2.39	2.54
TURBINE INLET TEMP	## # # # # # # # # # # # # # # # # # #	+145(°F)
OVERALL COMPRESSION RATIO	14.7	16.5
THRUST/WEIGHT RATIO	6.6	6.8





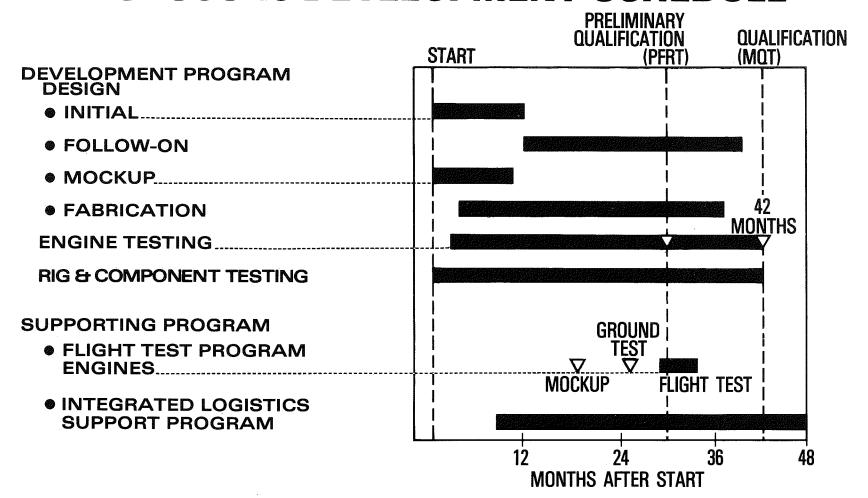
DEMONSTRATOR ENGINE TEST

- PEGASUS 15 SIZED FAN, RESTAGGERED
 COMPRESSOR, AND SHROUDED LOW PRESSURE TURBINE RUN ON PEGASUS 11 ENGINE
- ACHIEVED DESIRED FAN AIRFLOW
- DEMONSTRATED 15% THRUST INCREASE
 AT SEA LEVEL





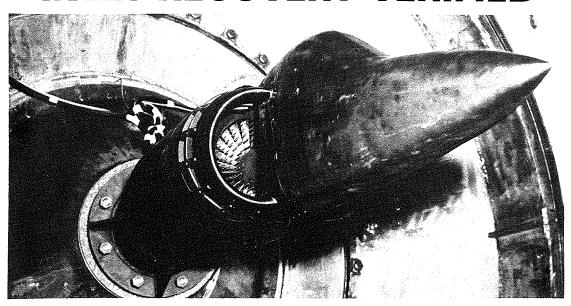
PEGASUS 15 DEVELOPMENT SCHEDULE







INLET RECOVERY VERIFIED

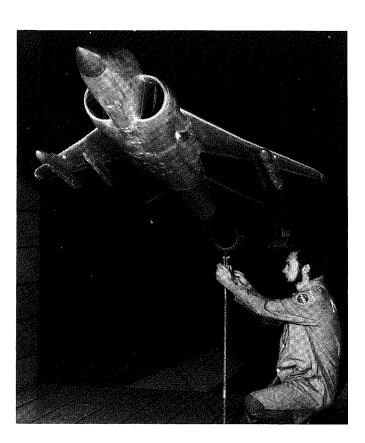


- NAVAL AIR PROPULSION TEST CENTER
- 300 RUNS (270 DATA POINTS PER RUN)
- ALL TEST CONFIGURATIONS EXCEEDED THE DESIGN GOAL OF 98% RECOVERY





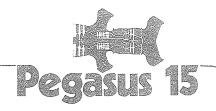
MISSION PERFORMANCE VERIFIED



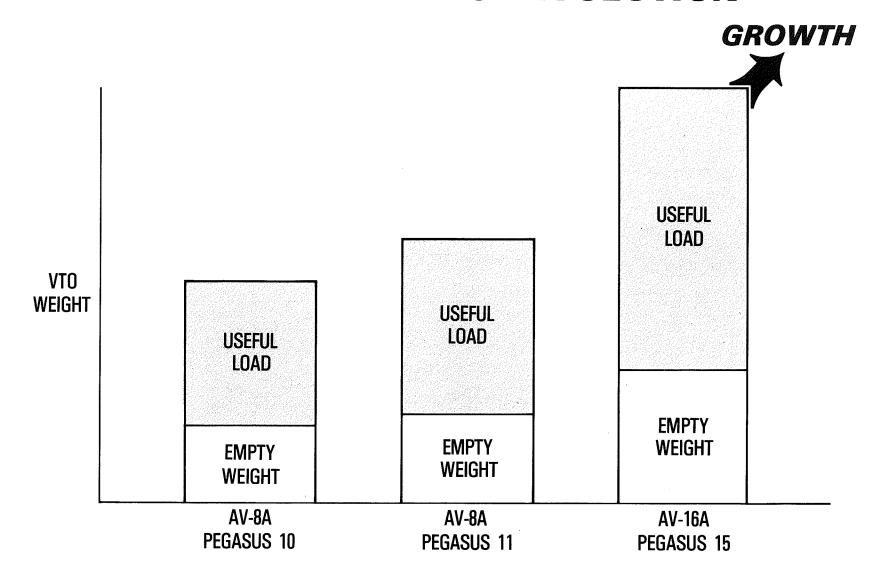
SUMMARY

- NASA AMES 11 FT. UNITARY TUNNEL
- 475 RUNS
- 276 OCCUPANCY HOURS
- THREE CONFIGURATIONS TESTED
 AV-8A
 AV-16A, W₂
 AV-16A, W₃
- EITHER ADVANCED TECHNOLOGY WING WILL MEET REQUIREMENTS.



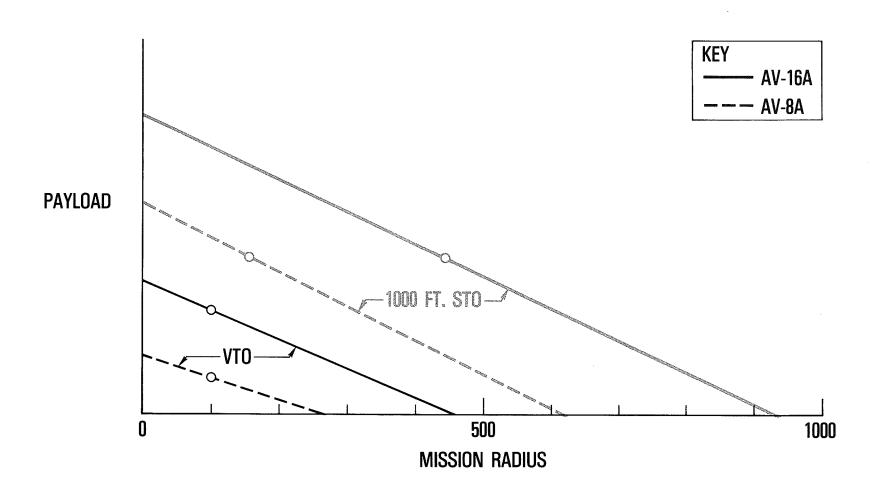


VTO PERFORMANCE EVOLUTION





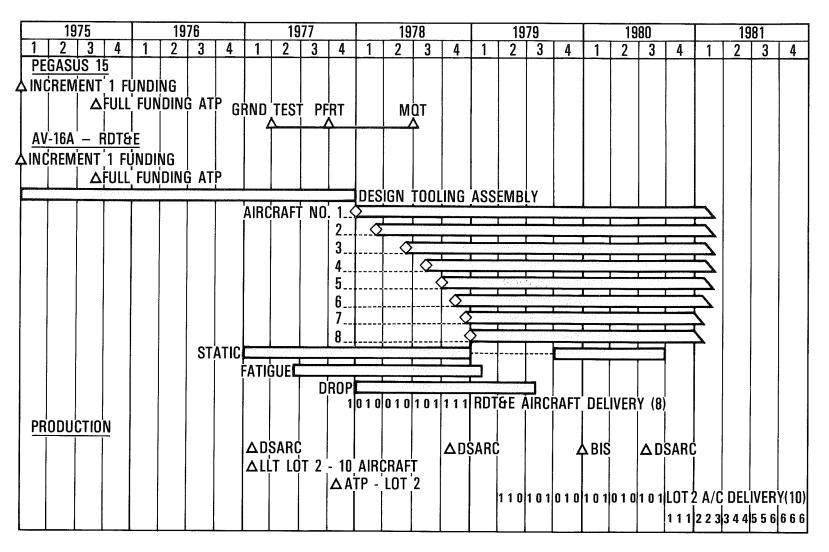
. . . . "GREATER THAN 2 TO 1 PERFORMANCE IMPROVEMENT"







AV-16A DEVELOPMENT AND PRODUCTION







AV-16A PROGRAM PLANNING PRICES

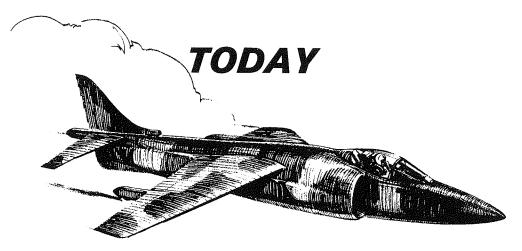
Millions of 1974 Dollars 350 Production Aircraft for USMC

AIRFRAME R&D	\$260.6
ENGINE DEVELOPMENT TO MQT	135.7
FLIGHT TEST ENGINES & SUPPORT	60.0
TOTAL	\$456.3 (1)
AVERAGE UNIT PRODUCTION	\$3.3

Notes: (1) Excludes Spares, Peculiar Ground Support Equipment and Navy Support Costs.







- ... "A Cost Effective Airplane Which Gives Substantial Performance Increases Over the AV-8A at Modest Cost"
- ON SCHEDULE
- WITHIN COST
- PEGASUS 15 DEFINED
- AIRCRAFT CONFIGURATION ESTABLISHED
- GREATER THAN 2 TO 1 PERFORMANCE IMPROVEMENT VERIFIED